

DEVELOPMENT OF SURFACE FRICTION ASSESSMENT MODEL AND MEASUREMENT FRAMEWORK FOR AIRPORT RUNWAY

A THESIS BY
PARIMAL KUMAR KARMAKAR

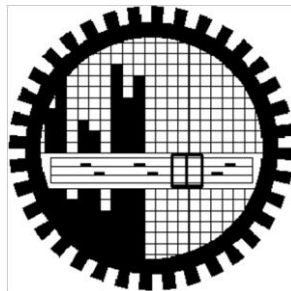
Submitted in Partial fulfillment of requirement for degree of
M.Sc. Engg.(Civil and Transportation)



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Dhaka , Bangladesh

March 2016
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**DEVELOPMENT OF SURFACE FRICTION ASSESSMENT
MODEL AND MEASUREMENT, FRAMEWORK FOR AIRPORT
RUNWAY.**



A THESIS SUBMITTED TO THE DEPARTMENT OF CIVIL ENGINEERING IN
PARTIAL FULFILLMENT OF THE REQUIREMENT FOR THE DEGREE OF

**MASTERS OF SCIENCE IN CIVIL ENGINEERING
(CIVIL AND TRANSPORTATION)**

**A THESIS SUBMITTED BY
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Dhaka, Bangladesh
MARCH, .2016**

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This thesis is dedicated to my parents Sri Paritosh Karmakar and Nilima Karmakar. Their blissful inspiration has made all my achievements possible.

ACKNOWLEDGEMENT

First of all, I would like to express my deepest sense of Gratitude to my supervisor Professor Dr. Md. Mizanur Rahman who offered his continuous advice and encouragement throughout the course of this thesis. I thank him for the systematic guidance and great effort he put into training me in the scientific field.

I would like to express my very sincere gratitude to Dr. Md. Hadiuzzaman, Assistant Professor, Department of Civil Engineering for the support to make this thesis possible.

I am thankful to the Manager of Shah Amanat International Airport, Wing Commander Nur-E Alam for the support and encouragement whenever I was in need. My gratitude also to Mr. Mats Jelvi of ASFT company of Sweden for his guidance during airport surface friction training programme.

Finally, I take this opportunity to express the profound gratitude from my deep heart to my beloved parents for their love and continuous support – both spiritually and materially.

ABSTRACT

International Civil Aviation Organization (ICAO) is the controlling authority of air traffic movement all over the world. Contracting states are given the authority to develop detailed schemes to provide acceptable levels of safety, both in respect of the objective and operational determination of surface friction . As a result, the methods of determination and availability of information differ widely between States. Although runway surface friction coefficient prediction models have been studied for more than half a century in the developed world, research on this topic in Bangladesh as well as in other south-east Asian countries has not been done frequently, indeed. This is mainly due to the complexity of data collection and processing.

Even though the current study has tried to focus some effort in this sector and develop a prediction model of runway surface friction coefficient in Bangladesh perspectives. There are many variables that are involved in this phenomenon including speed, ambient temperature, ground temperature, distance and inflation pressure of tire.

This study focuses on measurement framework and model development for runway surface friction coefficient based on data collected from Shah Amanat International Airport Runway. Friction data have been determined by Airport Surface Friction Tester (ASFT) car. The multiple linear regression equation has been developed from runway friction data at different speed, ambient temperature, ground temperature, distance and inflation pressure of wheel. This study outcome can be utilized for determination of friction coefficient of runway surface of other airports of our country, identification of level of safety of air traffic movement, evaluation of the effectiveness of aircraft performance, assessment of runway surface conditions and reporting a dissemination of runway surface conditions. This study has analyzed that cruise speed is more dominant over surface friction coefficient of runway surface. This study has revealed that an increment of 20% of speed value increases the friction coefficient value by 42% considering remaining parameters constant. On the other hand, increment of air temperature value by same amount increases friction coefficient value by 4.4%. Similarly, an increment of ground temperature, tire pressure and distance value by 20% result in a decrement of friction coefficient value by 8.8%, 32.6% and 2.9% respectively. The understanding of each of these variables will help one to realize the magnitude of the problem that one faces and eventually design more reliable friction measuring techniques.

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ACRONYMS AND ABBREVIATIONS

AGA	Aerodromes and Ground Aids
AIP	Aeronautical Information Publication
AIS	Aeronautical Information Services
ARC	Aviation Rulemaking Committee
ASFT	Airport Surface Friction Tester
ASM	Airport Services Manual
ATC	Air Traffic Control
ATS	Air Traffic Services
ATIS	Automatic Terminal Information Service
CAAB	Civil Aviation Authority of Bangladesh
CFME	Continuous Friction Measuring Equipment
CI	Confidence Interval
FAA	Federal Aviation Administration
FAR	Federal Aviation Regulations
GPS	Global Positioning System
GPRS	General Packet Radio Service
HMA	Hot Mix Asphalt
ICAO	International Civil Aviation Organization
IATA	International Air Transport Association

MPC	Mean Profile Depth
MPD	Measurement personal computer
NOTAM	Notice to Airman
NCHRP	National Cooperative Highway Research Program
PPC	Presentation Personal Computer
PDF	Portable Document Format
PFC	Porous Friction Course
RADAR	Radio Detection and Ranging
RSCSG	Runway Surface Condition Study Group
RPM	Revolutions per minute
SNOWTAM	Snow Notice to Airman
SARPS	Standards and Recommended Practices
SAS	Scandinavian Airlines System
SAIA	Shah Amanat International Airport
TALPA	Takeoff and Landing Performance Assessment
USA	United States of America
USB	Universal Serial Bus
WIFI	Wireless Fidelity

Chapter 1

INTRODUCTION

Air traffic communication does not have such a long history as road traffic or rail communication, yet the diversity of opinions related to the laws that govern friction is enormous. The purpose of this dissertation is to provide most up-to-date guidance on the subject of friction issues in aviation sector as far as is possible, given the present state of knowledge. It is widespread knowledge that pavements have a tendency to become slippery to both pedestrians and vehicles when they are wet or in undated or are covered with slush, snow or ice; however, no one yet has a holistic understanding of the physical effects causing this slipperiness which in turn can cause accidents. The same phenomenon applies to aircraft operation on the aircraft wheel movement areas. The information within this dissertation should be used by national authorities when implementing their flight safety activities and as necessary referenced for use by aerodrome operators, aerodrome air navigation service providers, aircraft operators and individuals within the organizations. This dissertation provides an overarching conceptual understanding of the surface friction characteristics contributing to controlling the aircraft via the critical tire ground contact area. Sufficient runway friction has a considerable impact on aircraft landing performance. This phenomenon is especially observed when aircrafts are landing on wet or otherwise contaminated runways due to the reduced braking action. With a view to preventing runway landings in excursion accidents and incidents, and enhancing airport and flight safety, available runway frictions should be studied. Therefore, runway surface friction assessments should be conducted following any significant maintenance activity conducted on the runway and before the runway is returned to service. Runway surface friction assessments should also be conducted following pilot reports of perceived poor braking action, if there are visible signs of a buildup of rubber deposits, runway surface wear, or for any other relevant reason.

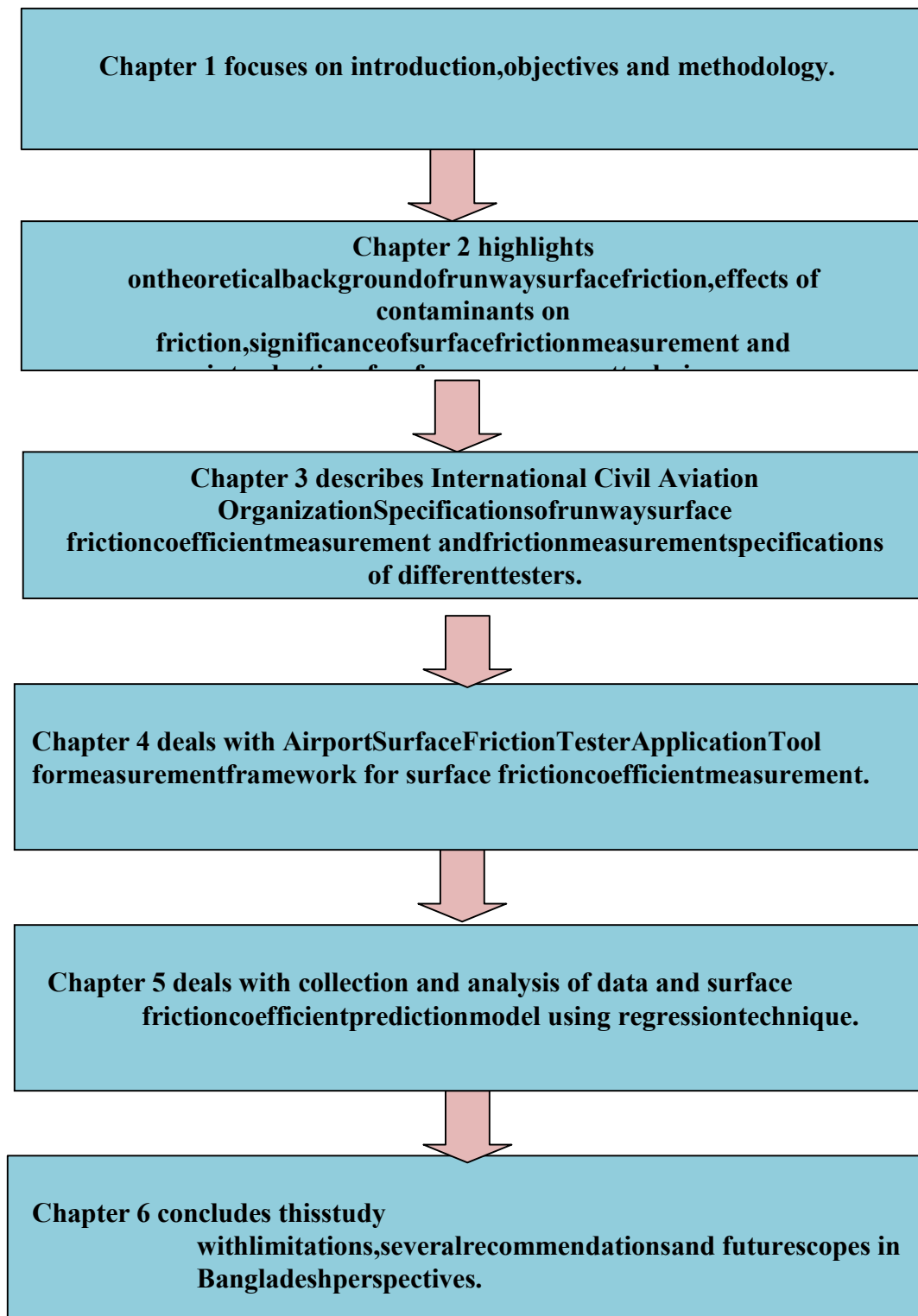
1.2 Objectives with Specific Aims and Possible Outcomes

This research is concerned with the determination of friction coefficient of runway surface of Shah Amanat International Airport, Chittagong. The objectives of this research are:

- a) To develop a comprehensive conceptual understanding of the surface friction characteristics contributing to controlling the aircraft via the critical tire to ground contact area.
- b) To calibrate the ASFT system and input the data into ASFT integrated software and measure coefficient of friction of Runway surface to suggest an applicable measurement framework following ICAO specifications.
- c) To develop a runway surface friction coefficient model incorporating local geo-environmental parameters.

The airport authority could use this information to assess the surface friction characteristics, evaluate aircraft performance considering interrelation among individual components of the systems such as, pavement surface (runway), tire (aircraft), contaminants (between the tire and the pavement) and atmosphere (temperature). These findings will lead and supplement numerous initiatives regarding establishment of rules and regulations of aviation safety, develop a system for reporting friction issues of the movement area as part of a standardized reporting format. This format must meet the needs of the pilot for safe operation of the aircraft; and develop a system for maintenance of the movement area.

1.3 Thesis Structure



Chapter 2

THEORETICAL BACKGROUND OF FRICTION

2.1 General Friction Theory

The simplest definition of friction is: The force F , required to tow an object, generating a certain normal reaction N , against a flat horizontal surface. Mathematically, the friction coefficient is expressed as below:

$$\mu(\text{mu}) = F / N \dots \dots \dots (1)$$

When friction is measured at airports according to ICAO procedures in Annex 14 and associated documents, the friction measured shall be the MAXIMUM FRICTION. Friction force is influenced by a combination of aircraft tires, aircraft braking systems, and airport runway pavement surfaces. The following listed factors are the main factors affecting runway friction.

- Ground Speed
- Slip Ratio;
- Tire texture and inflation pressure
- Pavement texture
- Water or contaminations.

2.1.1 The Classic Laws of Friction

The classic laws of friction evolved from the early work of Amontons and Coulomb. These laws were based on empirical observations and can be summarized as follows:

- (1) Friction is independent of the apparent or nominal contact area.
- (2) Friction force is proportional to the normal load.
- (3) Static coefficient of friction is greater than the kinetic coefficient of friction.
- (4) Kinetic friction is independent of the sliding speed.

The friction at the rubber tire-pavement interface constitutes a complex phenomenon due to the viscoelastic nature of the rubber. Empirical work conducted by many investigators shows that the classical laws of friction are not valid on viscoelastic materials. Denny (1953) conducted laboratory experiments on rubber-like materials and showed that under contaminated conditions the coefficient of friction decreases with increasing contact pressure. Thirion (1946) confirmed the load dependence of rubber friction and proposed an empirical relationship between the coefficient of friction and pressure. Schallamach (1952) showed that the load dependence of rubber friction can be explained by assuming spherical surface asperities and elastic behavior of rubber in compression. Although the mechanisms of tire-pavement friction interaction are not fully understood, the Molecular Attraction Theory, developed by Tomlinson and Hardy in the 1930s, seems to be the most accepted (Moore, 1975). The current investigation is limited to friction or skid resistance on wet pavement surfaces only.

2.1.2 Pavement Friction vs Tire Slip

This friction is measured by considering a certain slip as following equation

$$\mu = \frac{(\mu_{max} - \mu_{roll}) \cdot S + \mu_{roll}}{S} \dots \dots (2)$$

Slip, S, is generally expressed in percent. 100% slip stipulates that the braked wheel is skidding. Maximum μ (mu) is measured between 10 and 20% slip. Friction force has two components: the rolling resistance force and the slip resistance force. When the tire is rolling freely, only a rolling resistance force is applied on the wheel. As a braking force is applied, a slip occurs between the tire and the pavement surface. As shown in Figure 2.1, the tire proceeds from free rolling to a locked wheel, the coefficient of friction varies with the simultaneous changes of the tire slip. The coefficient of friction increases rapidly from a certain value, which is referred as rolling resistance coefficient, to a peak friction value and then it decreases to another certain value, which is referred as slider resistance coefficient.

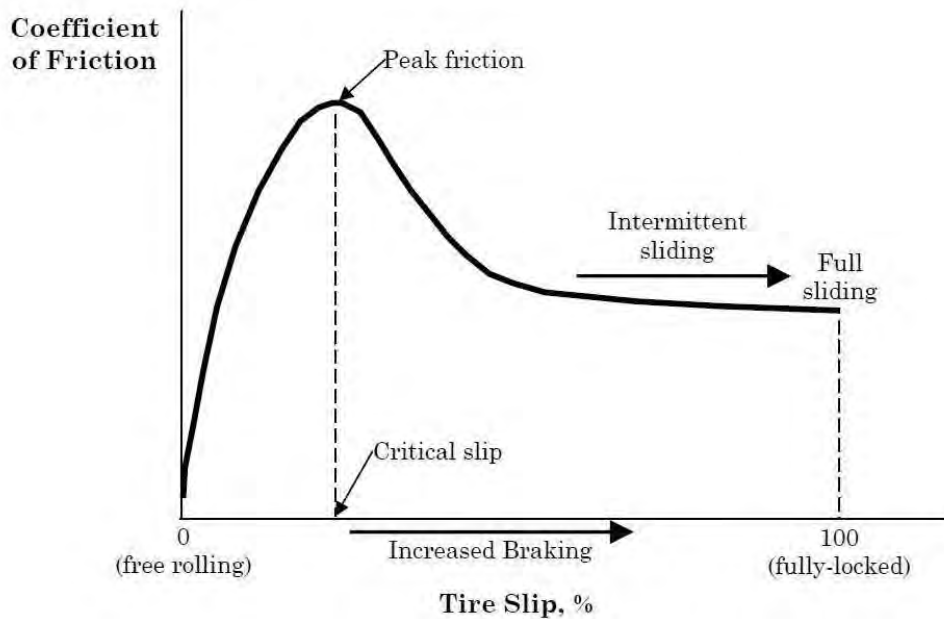


Figure 2.1 : Pavement Friction vs Tire Slip

The peak friction usually occurs when the tire slip exists between 10% to 20% slip, which is designated as the critical slip. When the slip proceeds to 100% slip, which refers to that wheel is completely locked, the coefficient decreases to a slider resistance friction coefficient. The slip resistance friction is lower than the peak friction. The difference between the slip resistance friction and peak friction is larger for wet and contaminated pavements in comparison with dry pavements. Friction characteristics of a runway depends on the structure of the runway surface. A good macrostructure should have a mean depth of about 1 mm. The micro-texture is depends on the aggregate used. For example, limestone will provide very low micro-texture and a runway will be very slippery, especially when wet. On the other hand, macrotexture refers to the overall texture of pavement. Figure 2.2 shows the microscopic view of microtexture and macrotexture of pavement. The speed has a significant effect on the friction of the runway surface when wet. If both macro-texture and micro-texture are poor the effect on friction of a runway when wet will be extremely dependent on speed. If both macro-texture and

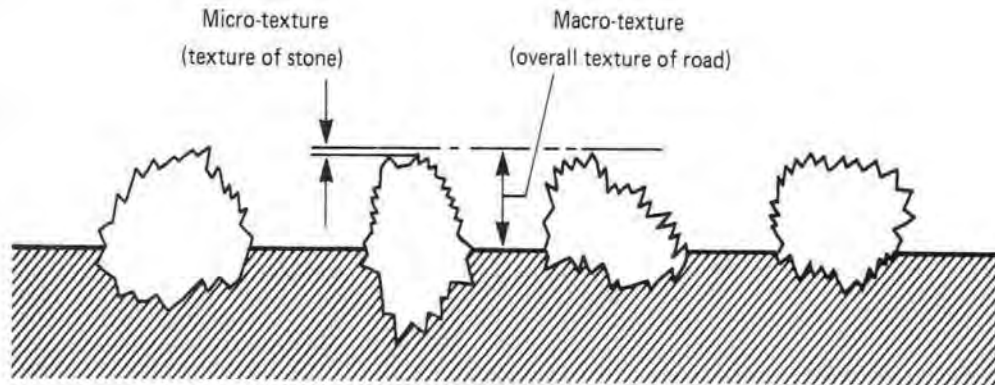


Figure 2.2: Micro-texture and Macro-texture

micro-

texture are poor the effect on friction of a runway when wet will be extremely dependent on speed.

Naturally, if only one of the texture kinds is poor the effect of speed on friction will be less consequently.

2.2 Vehicle Operational Parameters

2.2.1 Slip Ratio

Friction researchers use the slip ratio term to indicate the difference between tire velocity and vehicle velocity, as indicated in Equation (3).

$$Slip = \frac{v - \omega R}{v} \dots \dots \dots (3)$$

Where S is the velocity of the vehicle, ω is the angular velocity of the tire and R is the nominal radius of the tire. It is seen from Equation (10) that when the tire is rolling freely the slip must be 0 ($S = \omega R$). On the other hand when the tire is locked up the slip ratio is 1 ($\omega R = 0$). Locked wheels suffer severe localized wear under dry conditions since there is no rolling and subsequent uniform wear in the wheels when locked. Thus the material at the contact area between the wheel and the pavement surface is subjected to a frictional force that can lead to permanent deformation localized into one point only

of the wheel. On the other hand, a rolling wheel distributes these effects in a uniform manner throughout the circumference; therefore the wear is considerably lower than that in the locked wheel condition. Experimental work (NCHRP, 2009) shows that the maximum coefficient of friction for most surfaces is generally reached in a range between 0.1 and 0.2 slip ratio, depending on the type of surface, as shown in the Figure 2.3.

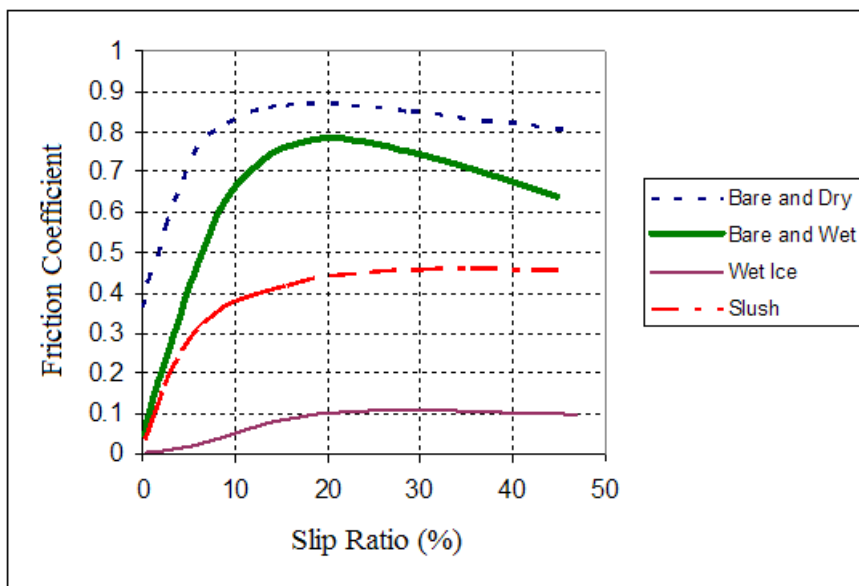


Figure 2.3: Coefficient of friction vs. Slip ratio on different surfaces (Adapted from NCHRP, 2009)

This is the principle on which Antilock Brake Systems (ABS) work. An ABS system recognizes that the maximum coefficient of friction is reached at a certain slip range, and hence controls the rotation of the tires for the slip ratio to be around that slip range.

Thus ABS prevents the tires from locking, which provides vehicle stability, steerability and improves stopping capabilities. The ABS is an independent system in that only the wheels that are about to be locked will be pumped and slip-controlled, while the others will be subjected to the full braking pressure. Consequently this system would allow one to stop a vehicle within the shortest possible distance. A computer monitors the speed of each wheel, which is fed into the ABS system (Mauer, 1995). When the system detects

that one or more tires have locked up or are returning relatively slower compared to the remaining tires, the computer sends a signal to momentarily remove and re-apply the braking pressure to the affected tire to allow it to continue turning. This "pumping" of the brakes occurs a ten or more times a second, far faster than a human can pump the brakes manually.

2.2.2 Vehicle Speed

In general, the friction coefficient decreases with speed on wet conditions. This phenomenon is attributed to the facilitation of drainage under the tire. The higher the speed, the less time the water under the patch has to drain off. Pavement macrotexture (MPD) is usually used to explain the friction-velocity dependency. High macrotexture improves the drainage properties of the tire patch area, avoiding hydroplaning conditions. Figure 3 shows both the effects of pavement macrotexture and vehicle speed on the coefficient of friction.

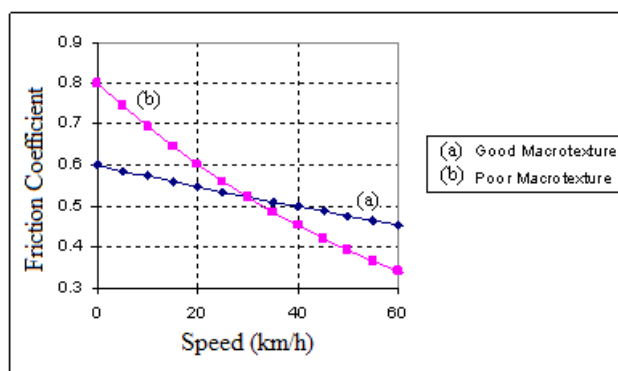


Figure 2.4: Texture Effect on Friction

2.2.3 Tire Characteristics

2.2.3.1 Tire Tread

The tire tread is a major factor when considering friction on contaminated pavement surfaces. Tire tread provides a drainage system to evacuate contaminants at the tire interface; thus having this a function as pavement macrotexture. The use of smooth

tires is recommended when performing friction tests on a pavement surface, because then information specific to drainage capabilities and texture of the pavement can be obtained.

2.2.3.2 Tire Inflation Pressure

Tire inflation pressure is directly related to the tire stiffness. Hence frictional characteristics of a tire are related directly to its inflation pressure. Low tire pressures will be reflected in higher rolling resistance. Figure 2.5 shows the results from a load-deflection test performed by the investigator using the smooth tire of the Locked Wheel Tester. It is seen that the tire stiffness increases with increasing inflation pressure.

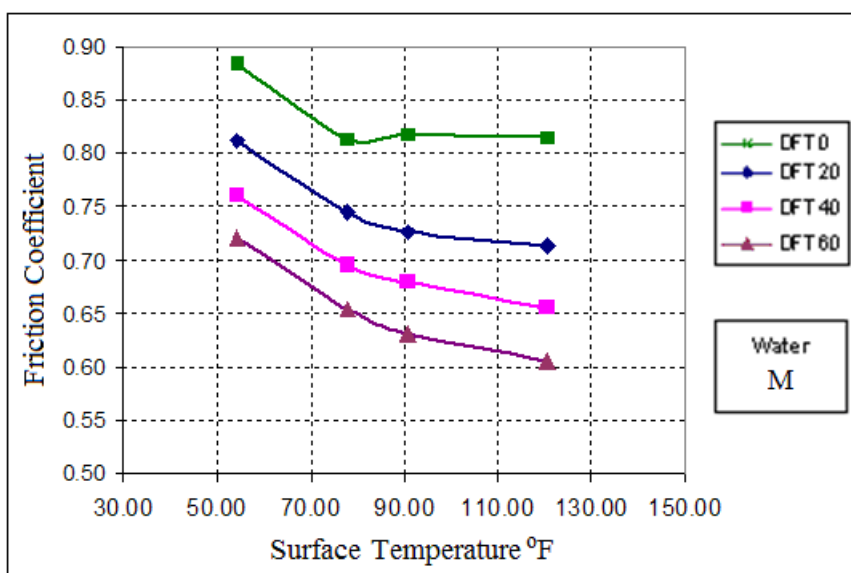


Figure 2.5: Friction Coefficient vs. Water Temperature in the range of 65 to 80°F (Medium)

From Figure 2.6 one can infer the effects that tire inflation pressure would have on the vertical load at the tire-pavement interface. As an example, a stiffer tire is more sensitive to a vertical displacement due to a profile than a softer tire.

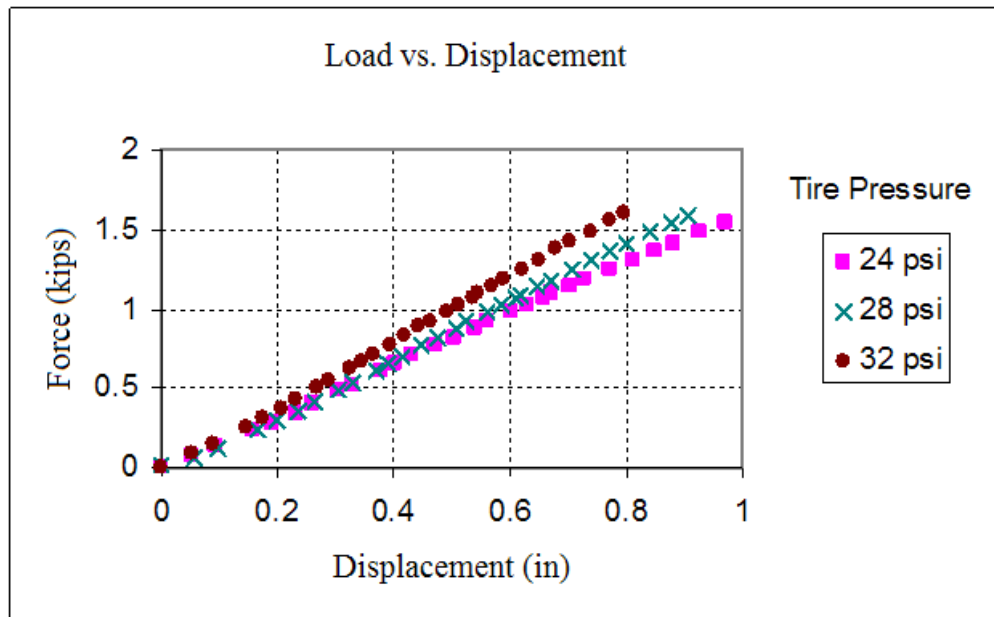


Figure 2.6: Effect of tire inflation pressure on tire stiffness.

2.2.4 Environmental Factors

A significant variation is observed in friction values measured on the same pavement surfaces at different times of the year. Several studies have suggested that this variation can be attributed to different environmental factors, such as rainfall, dry days preceding the measurement, temperature, cumulative vehicles passes on test lane; and greased deposits, etc (Jayawickrama et al., 1998).

2.2.4.1 Pavement Surface Temperature

Temperature has a significant effect on the frictional behavior of the tires, due to the viscoelastic nature of rubber. Friction in rubber-like material generally decreases with increasing temperature. Temperature effect on friction is the main parameter responsible for seasonal variations of friction measurements. Therefore, it is necessary to apply a temperature correction to friction measurements in order to perform comparisons between those at different temperatures or different seasons of the year (Fuentes et al., 2009).

2.3 Braking forces-general effects

Braking forces are generated by the friction between the tire and the runway surface during application of brake torque to the wheel. Friction generates when there is a relative speed between the wheels speed and the tire speed at the contact with the runway surface. This slip ratio is defined as the ratio between the brake and unbraked (zero slip) wheel rotation speeds in revolutions per minute (rpm). The maximum possible friction force is dependent mainly on the runway surface condition, the wheel load, the speed and the tire pressure. The maximum friction force occurs at the optimum slip ratio beyond which the friction decreases. The maximum braking force is dependent on the friction available as well as the braking system characteristics including anti-skid capability and torque capability. The maximum friction coefficient μ_{max} can exceed 0.6 on a good dry surface, which means that the braking force can represent more than 60 percent of the load on the brake wheel. On a dry runway, speed has little impact on μ_{max} . When the runway surface condition is degraded by contaminants such as water, rubber, slush, snow, or ice, μ_{max} can be reduced drastically, affecting the capability of the aircraft to decelerate after landing or during rejected takeoff. General effects of runway surface conditions on the braking friction coefficient can be briefly summarized as follows:

2.3.1 Wet condition (less than 3 mm water)

Friction Coefficient in wet conditions is much more affected by speed (decreasing when speed increases) in comparison with dry conditions. At a ground speed of 100 kts, μ_{max} on a wet runway with standard texture will be typically between 0.2 and 0.3, this is roughly half a fatal low speed such as 20 kts. On a wet runway, μ_{max} is also dependent on runway texture. A high micro texture (roughness) will develop the friction. A high macro texture, porous friction course (PFC) or surface grooving will enhance drainage benefits; however it should be depicted that the aircraft stopping performance will not be the same as on a dry runway. Conversely, runways polished by aircraft operations or

contaminated by rubber deposits or where texture is affected by rubber deposits after repeated operations can become very slippery. Therefore, maintenance must be performed periodically. There are several methods of removing rubber deposit from runway surface. The most common procedures include using chemical solvent, applying ultrahigh pressure water jet, mechanical grinding and shot blasting. Figure 2.6 shows the ultrahigh pressure jet vehicle and its effectiveness in removing rubber from runway.



**Figure 2.7(a): Before and after cleaning
high pressure by water jet**



**Figure 2.7(b): Ultra
water jet vehicle**



**Figure 2.7(c): Cleaning without
high pressure damaging surface**



**Figure 2.7(d): Ultra
water jet**

2.3.2 Loose contaminants (standing water, slush, wet or dry snow above 3mm)

These contaminants degrade μ_{\max} to levels which could be expected to be less than half of those experienced on a wet runway. Microtexture has little effect in these conditions. Snow results in a fairly constant μ_{\max} with velocity, while slush and standing water exhibit a significant effect of velocity on μ_{\max} . Water and slush create dynamic aquaplaning at high speeds because they have a fluidic behaviour, a phenomenon where the fluid's dynamic pressure exceeds the tire pressure and forces the fluid between the tire and ground, effectively preventing physical contact between them. Figure 2.7 shows the phenomenon of aquaplaning on a moving wheel. In these conditions, the braking capability drops drastically, approaching/reaching nil.

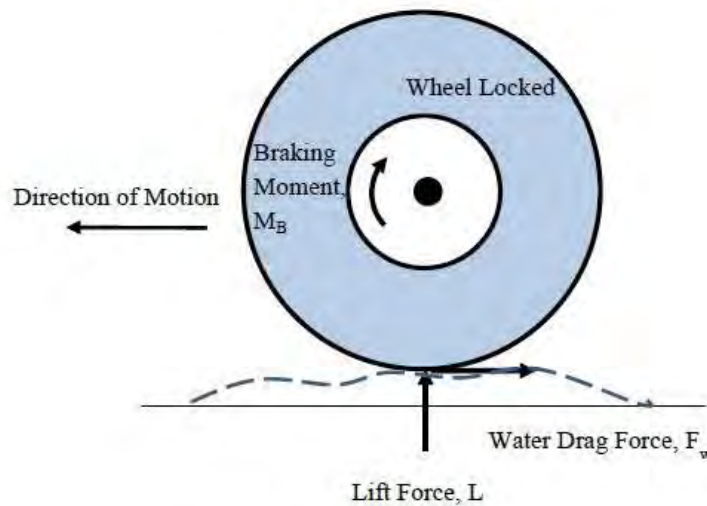


Figure 2.8: An Aquaplaning Landing Gear Wheel on Wet Runway Pavement.

2.2.3 Solid contaminants (compacted snow, ice and rubber)

These contaminants affect the deceleration capability of aircraft by reducing μ_{\max} during landing and rejected taking off. These contaminants have no effect on acceleration. Compacted snow may exhibit quite good friction characteristics, perhaps comparing with a wet runway. However, when the surface temperature approaches or exceeds 0°C , compacted snow will become more slippery, potentially reaching a very low μ_{\max} . The stopping capability on ice can vary depending on the temperature and roughness of the

surface. Generally, wet ice has shows very low friction (μ_{\max} dropping to as low as 0.05) and will typically prevent normal aircraft operations until the friction level is improved. However, ice that is not melting may still allow operations, albeit with a performance penalty.

2.4 Reasons of measurement of friction of runway

Flight Safety is the main reason for measuring friction of runway surface. It has become more important to check friction in a better way than making skid tests. Scandinavia, particularly Sweden, has taken a considerable part in the development of friction measuring technique. Among reasons for friction measurements are:

- Determine friction characteristics of runways under winter conditions
- Verify friction characteristics of new or resurfaced runways
- Assess periodically the slipperiness of paved runways when wet
- Assess the effect on friction when drainage characteristics are poor
- Assess friction of runways becoming slippery under unusual conditions

2.4.1 Early reporting technique.

The early reporting technique was developed in joint cooperation of the Airport Authority at Bromma Airport and SAS. This took place in the early 1950. This led to reporting friction characteristics for three parts of the runway seen in the direction of landing. Soon the thirds were called A, B, and C. A is always called the low number runway end. An aeroplanes landing from the high number direction got the report on friction in the order C, B, and A. SAS and domestic Swedish operators understood what the friction numbers meant to them. However, operators coming into e.g. Bromma airport did not understand what the reported numbers meant. Therefore, the expressions Good, Medium, and Poor were introduced. SAS sent out a questionnaire asking for information from pilots on how they experienced information on braking action, i.e.

friction, and also on controllability in crosswind. About 3000 answers on this questionnaire were received. The answers demonstrated that when a friction coefficient of 0.40 or above had been reported there were no pronounced problems on braking or controllability in crosswind. When 0.25 or lower had been reported the problems became severe. As a result of this study in Sweden was introduced the terminology:

Table 2.1 : Reporting of Pavement Surface Condition

Good	0.40 and above
Medium to Good	0.36 to 0.39
Medium	0.30 to 0.35
Medium to Poor	0.26 to 0.29
Poor	0.25 and below

As can be seen from the above range of friction coefficients, we consider that no more than two significant figures should be reported. More than two figures would give a false impression of accuracy of the friction measuring equipment.

2.5 Terminology

Micro texture

Micro texture is the 'fine scale roughness' contributed by small individual aggregate particles which is detectable by touch rather than appearance. It allows the tire to break through the residual water film that remains when the bulk of water has run off and is especially important at low speeds.

Macrot texture

Macrot texture is "visible roughness" and allows water to escape from beneath aircraft tires. It becomes more important as the factors which can lead to aquaplaning come into play - increasing speed, decreasing tire tread depth and increasing water depth.

Aquaplaning

Aquaplaning or hydroplaning by the tires of a road vehicle or aircraft occurs when a layer of water builds between the wheels of the vehicle and the road surface, leading to a loss of traction that prevents the vehicle from responding to control inputs.

Contaminant

A deposit (such as snow, slush, ice, standing water, mud, dust, sand, oil, and rubber) on an aerodrome pavement which is detrimental to the friction characteristics of the pavement surface.

Braking action

A term used by pilots to characterize the deceleration associated with the wheel braking effort and directional controllability of the aircraft.

Surface friction characteristics.

The physical, functional and operational features or attributes of friction that relate to the surface properties of the pavement that can be distinguished from each other.

Chapter 3

SURFACE FRICTION OF AIRPORT RUNWAY

3.1 General

International Air Transport Association, IATA, arranged a conference in 1952 where SAS was given the prospect to signify the Scandinavian experience about knowing the friction characteristic of runway surface and the procedures for measuring and reporting friction characteristics at airports. Discussions dealt only with operational measurements as the inevitability for measuring of friction on wet runway had not yet been documented. The findings of the meeting was that IATA formulated that there is an operational demand for consistent and standardized information concerning the friction characteristic of ice and snow covered runways. At ICAO fifth Aerodromes and Ground Aids Divisional meeting, 1952, AGA5, IATA forwarded its conclusion. The conclusion made by IATA was established and is still found in the existing valid edition of ICAO Annex 14, Aerodromes, July 1995. The ICAO effort on development of specifications on operational friction measurements at airports that was commenced at the AGA5th divisional meeting 1952 has continued over the years and gradually lead to the specifications now found in Annex 14.

3.2 ICAO Annex 14 regarding operational surface friction coefficient

The intent of these specifications is to satisfy the SNOWTAM and NOTAM promulgation requirements. It is recommended that whenever a runway is affected by snow, snow or ice, and it has not been possible to clear the precipitant completely, the condition of the runway should be assessed, and the friction coefficient measured. The readings of the friction measuring device on snow, slush or ice-covered surfaces should adequately correlate with the readings of one or other such device. The primary aim is to measure surface friction in a manner that is relevant to the friction experienced by an

aircraft tire, thereby providing correlation between the friction measuring device and aircraft braking performance.

3.3 Accidents and Incidents Leading to New Rules

An air traffic accident in the 1970, at Los Angeles International Airport conspicuously demonstrated that this accident would not have occurred, if a system of periodical measurements of friction characteristics of runways when wet had been in use. The accident resulted in complete loss of a Jumbo-jet DC-

10. The accident investigation showed that the touchdown zone was extremely slippery under wet conditions. The accident would not have occurred, if the unsatisfactory friction characteristics had been known and corrective measures taken. It happened also a lot of other accidents and incidents, where slipperiness of runways when wet had been a conducive factor. A similar aircraft accident took place on the 8th October, 2004 at Osmani International Airport. An AF28 aircraft of Bangladesh Biman skidded off the runway while landing in heavy downpour shown in figure 3.1.



Figure 3.1: Aircraft Accident at Osmani International Airport, Sylhet.

ICAO commenced work on including specifications related to friction measurements on runway when wet in Annex 14. For the AGA 8th Divisional meeting 1981 a special

working group, the Runway Surface Condition Study Group, RSCSG, prepared proposals for material to be included in Annex 14.

3.4 Friction measuring technique on wet runways

When studying friction measuring technique on runways when wet, it was soon observed that the measuring equipment had to make a continuous measurement of friction. It also had to have capability of self-watering the runway. When discussing the self-watering capability within the ICAO RSCSG, it was argued that this requirement was not necessary as in most countries it rains so often that artificial watering was not necessary. However, if depending on natural rain one would not know what the water depth during the friction measurement. This argument was accepted and self-watering features became an agreed requirement.

3.5 ICAO Annex 14 Regarding Surface Friction Coefficient Measurements

- Deals with friction on new or resurfaced runways. The friction tests should be made on a clean surface. Especially on new surfaces, or resurfaced runways, an aerodrome operator should carry out additional friction testing to establish friction readings during adverse weather conditions and to identify those areas of the runway where contamination (i.e. water) may build up over a short period of time. This is of particular importance where reprofiling of the runway's lateral, longitudinal or sloping planes has been accomplished as part of any rehabilitation project.
- Draws attention to the friction on runways during wet conditions and should be checked periodically in order to identify runways having unsatisfactory friction characteristics when wet.
- Depicts that States should define what minimum friction level it considers acceptable. The minimum friction values should be published in the Aeronautical

Information Publication, AIP. When the friction found is below this value a NOTAM should be issued and pilots should be informed that the runway surface may be slippery when wet. When friction is below this level corrective actions should be initiated. However, when the friction level found is found to be below minimum level corrective action must be taken without delay.

- Provides guidelines on time interval of measurement. Friction measurements should be taken at intervals that will ensure identification of runways in need of maintenance before the condition becomes serious, or that the friction of the runway when wet has not fallen below the minimum friction levels specified by the State. The time interval between measurements will depend on factors such as: aircraft type and frequency of usage, climatic conditions, pavement type, and pavement service and maintenance requirements. Table 3.1 shows recommended maximum interval between runway surface friction assessments.

Table 3.1 Recommended Maximum Interval Between Runway Surface Friction Assessments.

Average number of movements on the Runway per day	Maximum Interval between Assessment
Less than 400	11 months
400 or more	5 months

- Depicts about tire category of different friction measuring vehicles. Different types of friction measuring devices have been shown in figure 3.2. The two friction measuring tires mounted on the μ (mu) meter were smooth tread and had a special rubber compound, i.e. Type A. The single friction measuring tires mounted on the Skidometer, Surface Friction Tester, Runway Friction Tester and Tatra were smooth tread and used the same rubber compound, i.e. Type B.

The Griptester had a similar tire but the size was smaller, i.e. Type C. Table 3.2 shows the specifications of different friction testers.

Table 3.2: Specifications of Different Friction Testers.

Test Equipment	Tire Type	Tire Pressure (bar)	Test speed (Km/h)	Water depth (mm)	New surface	Maintenance planning	Minimum friction level
Mu-meter trailer	A	70	65	1.0	0.72	0.52	0.42
	A	70	95	1.0	0.66	0.38	0.36
Skidometer Trailer	B	210	65	1.0	0.82	0.60	0.50
	B	210	95	1.0	0.74	0.60	0.34
Surface Friction Tester	B	210	65	1.0	0.82	0.60	0.50
	B	210	95	1.0	0.74	0.47	0.34
Runway Friction Tester Vehicle	B	210	65	1.0	0.82	0.54	0.50
	B	210	95	1.0	0.74	0.54	0.41
TATRA Friction Tester Vehicle	B	210	65	1.0	0.76	0.54	0.48
	B	210	95	1.0	0.67	0.57	0.42
GRIPTESTER Trailer	C	140	65	1.0	0.74	0.57	0.48
	C	140	95	1.0	0.64	0.36	0.24



Locked wheels smooth/ribbed



Griptester (GT)



Skid Wheel Testers



Dynamic Friction Tester

Figure 3.2: Different types of Friction Measuring Devices.

Chapter 4

SURFACE FRICTION MEASUREMENT FRAMEWORK BY ASFT TOOL

4.1 General

When the system starts up, the driver has to login with a unique ID and password as shown in Figure 4.1. The type of tire, tire pressure and water film level should also be input there. These login values are used as a confirmation from the user that the measuring vehicle has been checked for the correct measuring tire and pressure.



Figure 4.1: User ID popup menu.

There are four levels of security for the users: Measurer, Calibrator, Configurer and Superuser.

- Measurer (MEAS), can measure and test calibration accuracy
- Calibrator (CALIB), has the same rights as measurer and can calibrate the system.

- Configurer (RWY), has the same rights as measurer and calibrator and can define runways (measure objects), delete runways, and can erase measurements from the system when doing backups.
- Superuser (SUPER), service personnel level with access to some functions for system diagnose.

4.2 Main Menu (Measure tab)

All interaction with the measuring system is done via the touchscreen on the PPC as shown in Figure 4.2.

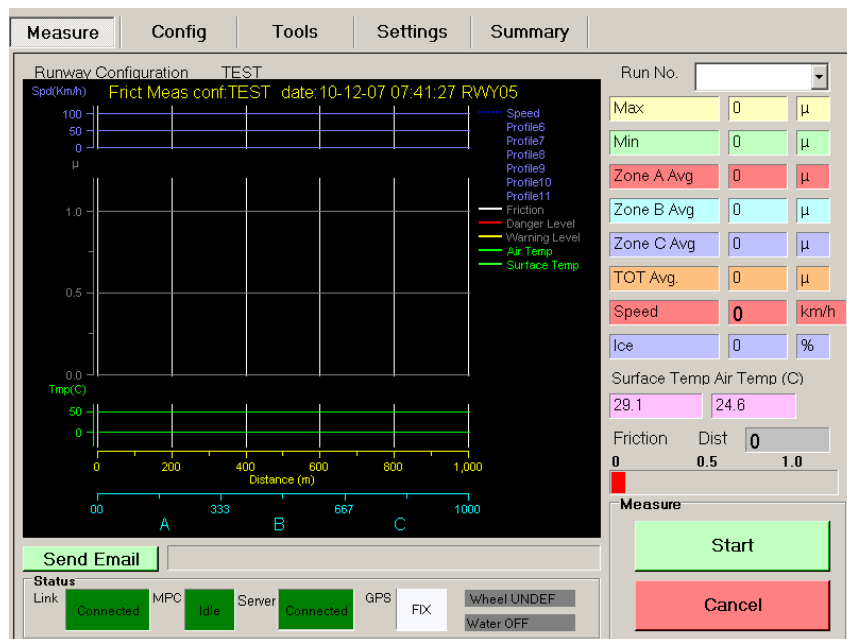


Figure 4.2: Main menu.

This tab is the view of the PPC that is used to operate the equipment. From this view the driver will start, stop and supervise measuring.

Link

When the system is started and the PPC has established a connection with the MPC this status box will output “Connected” and will appear green, else it will read “Ready to Connect” in red.

MPC

This box demonstrates the status of the MPC, normally if the system is not measuring this will demonstrate “Idle” on a green background.

Server

This box shows the status of the connection to the ASFT Measure Server, it could be connected (green colour) or not connected (red colour). To connect to the ASFT server this requires an Internet connection, through GPRS, WIFI or other.

GPS

This box tells the driver if the GPS antenna has picked up the position of the vehicle. This is then displayed in white with the text “Fix”.

Wheel /Water status

When the system is started the MPC sends the actual status of the wheel (undefined, up, down or pumping) and the water (on or off).

Measure Graph

Friction, air temperature, surface temperature and speed will appear here as three separate graphs (the temperatures will be combined into one graph) in real time as a measure is performed.

Send Email

This button will send all measure files and pdf reports that has not previously been sent, to two predefined email addresses.

Message area

Various messages from the system will end up here, it can be e.g. WaterTankEmpty.

Run No

Shows the current run number, useful if the equipment is used for multi runs.

Max, Min

Shows the maximum/minimum friction value for the current measure.

Zone A Avg, Zone B Avg, Zone C, TOT Avg.

Shows the average friction for A, B, C zone and the total average of the current measure.

Speed

Shows the current speed of the vehicle.

Ice

Shows the runway ice percentage of the current measure (what is regarded as ice is configured for the runway in the Configuration tab)

Surface Temp/Air Temp(C)

Shows the current temperature for air and surface if there is a temperature sensor installed.

Dist

Shows actual distance measured.

Friction

Bar showing the current friction value graphically in color coded squares.

Start

This button starts a measure according to the current settings in Config tab. When the system is started the measuring wheel will be lowered and an instruction window will tell the user what to do.

Cancel

This button will cancel an ongoing measure when the user for any reason wants to stop it.

4.3 MapMenu(Map tab)

This menu will only be available when the GPS antenna is connected. When the GPS antenna is connected and a map for the area is loaded, every point of friction is plotted on the map as the measure is conducted. The measured temperature is also plotted here if a temperature device is connected as shown in Figure 4.3.

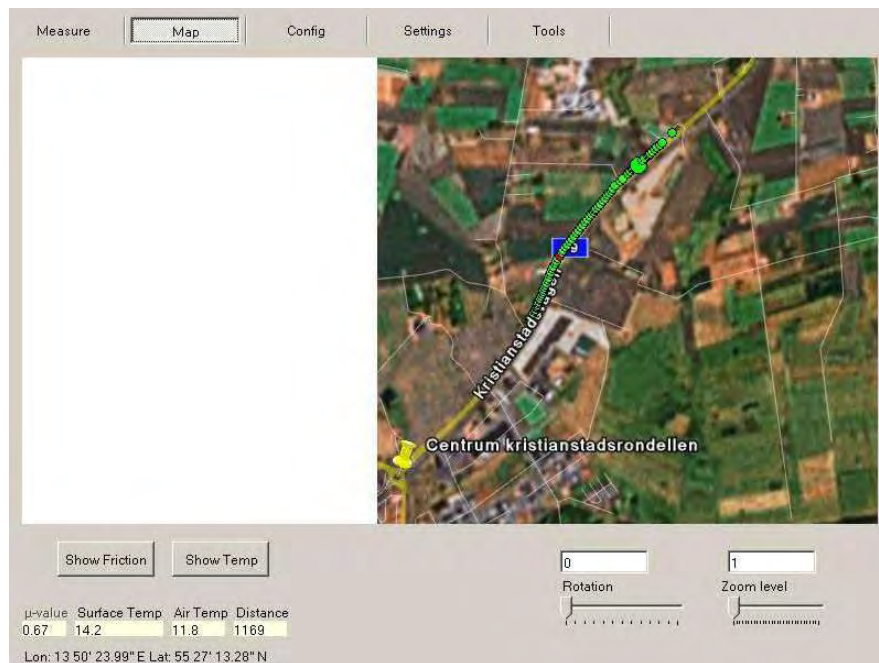


Figure 4.3: MapMenu(Map tab).

Map

This tab shows the map of the area that is measured (configured by ASFT personnel, in the settings file). All measure values will be presented here as round dots if the GPS

antennais engaged during the measure. The friction will be indicated here as red, yellow or green dots according to the levels that are configured in the system.

Show friction

If the temperature values are recurrently visible in the map area this button will switch to friction values.

Show temperature

If the friction values are recurrently visible the view will change to show the temperature values.

Single dot data

When a single measured value is selected, the data for that position is displayed in the lower part of the screen. The friction, surface and air and temperature, speed and distance from origin is displayed here together with the positions latitude and longitude.

Rotation and magnification of map

The sliders in the bottom of the view controls the rotation angle and zoom level of the map.

4.4.4 Configuration menu (Config Tab)

This tab is used for configuration of measuring objects e.g. runways or road fragments. Here are real utilities for technicians to check different parts of the system as showed in Figure 4.4.

Runway Data Group

Name

All runways have a user chosen name that is defined in this text field

Speed

This is the cruise speed for the configured runway. When this speed has been measured, a signal will sound so that the driver will know that the cruising speed has been reached.

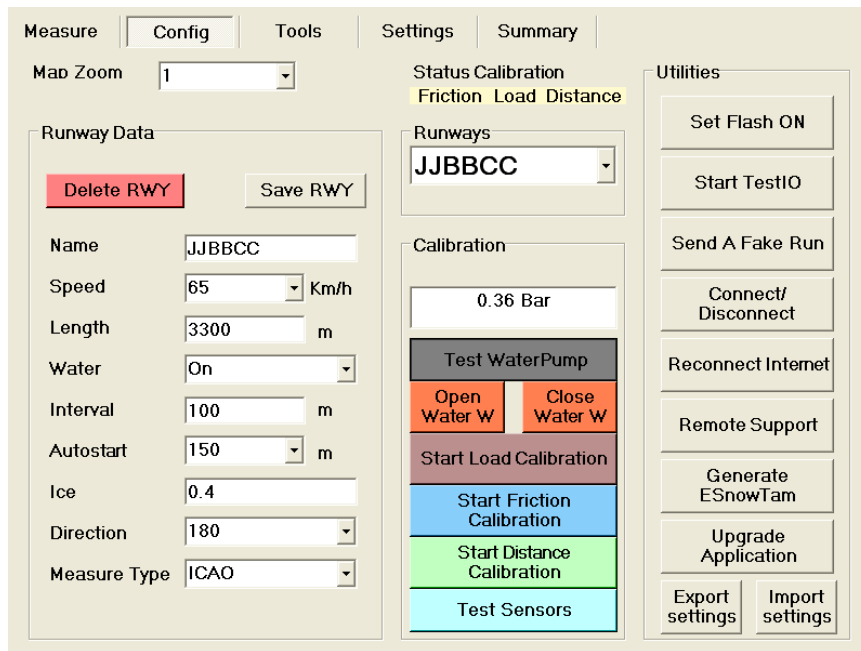


Figure 4.4: Configuration menu (Config Tab).

Length

This is the length of the distance to be measured in meters. It is noted that this is not the complete length of the runway/road, only the part to be measured. Distance for acceleration and braking should be allowed.

Water

This configuration tells if water shall be engaged during the measuring, On, Off, Auto.

Interval

This configuration tells on what distance the average of the friction value is calculated in the tools data table view.

Autostart

This configures the length in meters that the vehicle should move before the actual measuring starts. (0 = Manual start)

Ice level

This tells what friction level that will be considered as ice, used in the measure tab.

Direction

This tells what direction the measuring starts in, this parameter determines where the AB and C zones are regarded to be, The A zone is always considered to be in the direction where the degree value is below 180 degrees.

Measure Type

This configures what type of measure to do, normal maintenance measures are of ICAO type.

Save / Delete RWY button

It is the button to save the Runway data to file. When a new runway is configured, press this button and the data is saved and ready to use. If the user wants to delete a runway, he can choose the runway to be deleted from the drop-down list and press "Delete RWY".

Runways

Drop-down list provides an option to choose and open a predefined runway data file.

Default Map Zoom

This configures the startup map zoom level.

Calibration status

These labels indicate the calibration status. A red label indicates that the calibration for that function has not been performed.

Water Pressure

This shows current water pressure when measuring. This window is used for setting the pressure when not using the electrical watering system.

Test Pump button

The user can start the electric pump by pressing this for a few seconds to test that it is working properly. When using this watering system the user doesn't need to calibrate. The user can set the water film during logging and then configure the water to Auto. The pressure will be automatically adjusted.

Open / Close Water Valve

During a measure, the user can use these buttons to set the pressure when not using the electrical watering system.

Start Load Calibration

This starts the calibration of the load cell.

Start Friction Calibration

This starts the calibration of the friction cell.

Start Distance Calibration

This starts the calibration of the distance.

Check Calibration

The Check Calibration button is to test if the load and friction calibration was performed correctly and also gives the possibility to see if the pulse transmitter (distance) works.

Special buttons

The only button available for regular users is the "Remote Support" button. This is used to make ASFT service personnel able to help with problems via Internet. This is possible only on equipment with a connection to Internet, e.g. a GPRS modem or a link to a wireless network (WIFI).

Set Flash On

This is used to set the flashing light on if you have a trailer (T5 or T10).

Start Test IO

This is used by authorized service personnel to check the system.

Send A Fake Run

This sends measurements to the ASFT server to check that the connection is working properly. It is needed to press the button a second time to stop the sending of measure values.

Connect/Disconnect

This creates a new log into the server.

Reconnect Internet

This is used only for older modems where it disconnects and then reconnects with user and password authentication.

Remote Support

This starts a remote support program, Team Viewer, in a new window. This requires access to Internet (Option). The user can get an ID and password that ASFT needs to connect to PPC. From this connection ASFT can support PPC as well as MPC.

Generate Snowtam

This starts a view where the user can create a SNOWTAM report. Measured values will be automatically filled. This report can then be sent by e-mail.

Upgrade Application

If the user chooses to upgrade application with e.g. Snowtam or External applications, this is used to transfer the software to PPC.

Export Settings

It's recommended to use this to make a security backup of system files. This will copy your runways, settings, ports, users, maps and other system files to be able to recreate your system. They will all be saved to a USB memory stick.

Import Settings

This is used to import all your system settings.

4.5 5 Settingsmenu(Settings Tab)

This tab is used for the settings as showed in figure 4.5. The only values the user can set are the warning and danger levels. Otherwise this tab is only for showing the settings of the system.

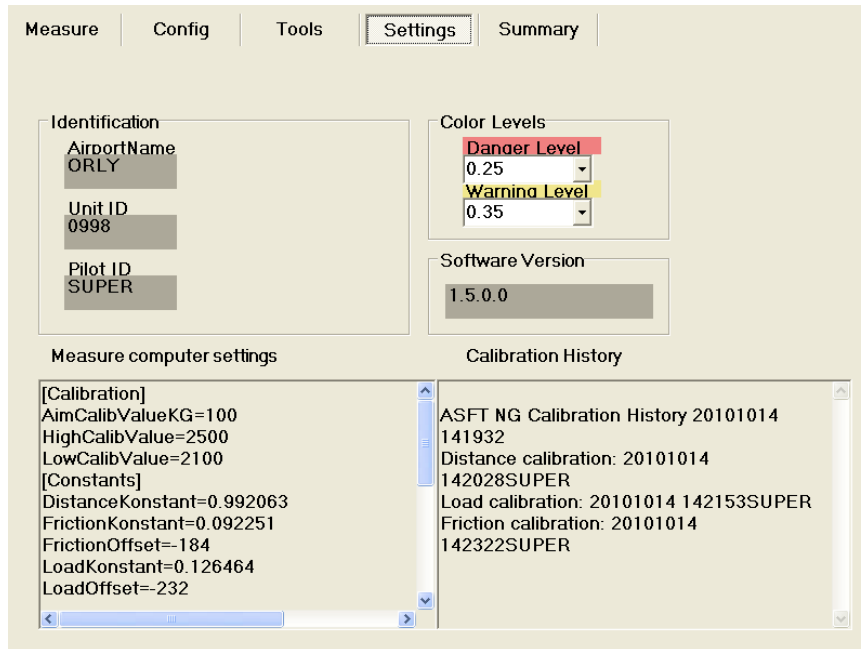


Figure 4.5: Settingsmenu(Settings Tab)

Airport Name

Shows the airport name for the equipment as defined in the settings file.

Unit ID

The Unique ID for equipment. The SFT Number is defined in the settings file as well as imprinted in the vehicle.

Pilot ID

This shows the ID of the person logged in.

Color Levels

Shows the threshold friction values for the warning and danger level in the system.

Software Version

This software revision of the PPC is displayed here.

Measure Computer Settings

This area displays the settings, calibration and software revision values for the measure computer, this can be used for troubleshooting issues.

Calibration History

This area displays the calibration history. Date and time, user ID and type of calibration done is displayed here.

4.6 6 Tools Menu (Tools Tab)

The tools menu is a tab where it is possible to load old measurements for examination as shown in figure 4.6. It is also possible to view the data in tabular format and do backup to memory stick.

Run No

This shows which run that is displayed in the data area and gives the possibility to change run to view.

Run Data

This shows a collection of data from the run way and the displayed run.

Friction Measure Tabular Data

This shows the current measurement data in tabular format. A click in the average column will recalculate averages with the distance of that row as base

MPC Connection status

This button shows the current status of the MPC.

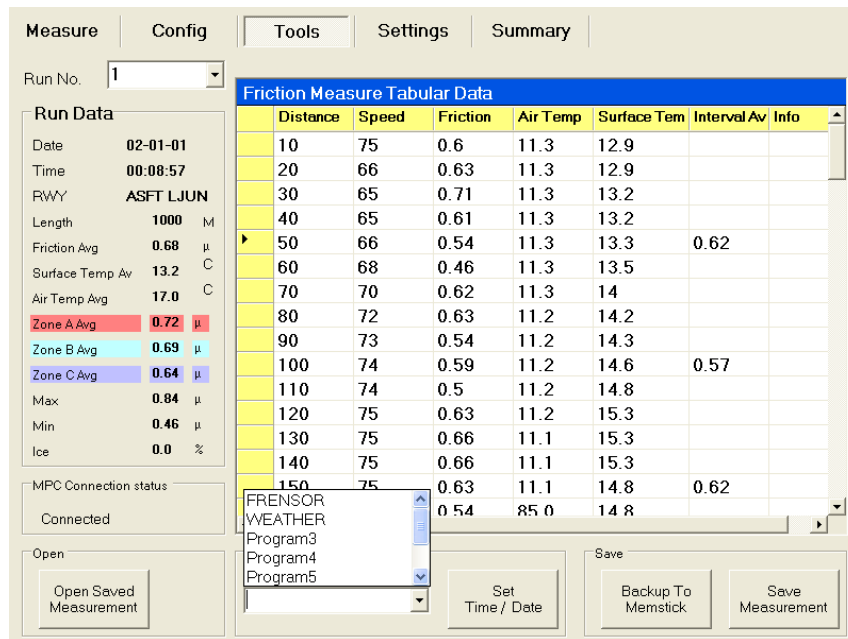


Figure 4.6: Tools Menu (Tools Tab)

Open Saved Measurement

This is the button to open an old, previously saved measurement with.

External Application

This gives the possibility to open another application.

Open / Print PDF Report

This opens a Adobe PDF reader window where the user can report in the car.

Backup To Memstick

Backup of all measured data to a USB memory stick.

Save Measurement button

Saves the current measurement.

4.7 Start up the system

From the time that the system is started it takes one minute before the system is ready to measure, so it's a good idea to start the system as soon as the car engine is started. The system is ready to measure when the status of the Link and MPC shows up like this.

Trailer(T5, T10)

On the trailer the user has to start the PPC and MPC independently, it doesn't matter what unit is started first as long as both systems are restarted. The start button for the MPC is placed in the Trailer, set the switch to 1. The PPC is started with the power button on the bottom of the screen.

Car

The car has one switch that starts the whole system. It's placed in the car's center console.

4.8 8 Shut down the system

Trailer(T5, T10)

When the switch in the trailer is set to 0 the system will shut down in ~30 seconds. When the system has shut down you need to turn off the power to the PPC.

Car

The same switch is used to start the system is used to turn the system off, no other action is called for, the system has to be off for a minimum of two minutes before it is ready to be started again the power.

4.9 9 Calibration of the system

The system has to be calibrated before it can be used. It is of big importance that this is done with accuracy, an incorrectly calibrated system will never give any trustworthy results.

Scales Floor

r scale

The floor mounted scale can be used with a power cord (usable for 100–240V) or three LR14 batteries.

Top mounted scale

This scale uses three AA batteries. Both scales should turn off automatically after a few moments.

Calibration of vertical load cell (red lane)

It is important that the vehicle has the same weight throughout the entire calibration process. This means that the user cannot fill the water-tank whilst calibrating nor can users alternate between sitting in the car or being on the outside. It's recommended to have two users doing the calibration, one reading the scale and one operating the PPC, or one user being on the outside of the car at all times, the user has to lean in and operate the PPC.

Calibration of water pressure (orange lane)

Calibration of water pressure is only intended for self-watering systems. The actual pressure is presented in the pressure display. This display is in the config mode just above open/close water valve buttons. The calibration has to be performed in the desired speed to use when measuring. For safety reasons it's recommended to be two operators, one to calibrate and one to drive and hold the speed. It is a hazard to keep track of the car, speed, pressure and operations simultaneously. The pressure should be adjusted according to this table 4.1.

Table 4.1: Pressure Adjustment Table

Film thickness	Pressure (Bar)	Speed
1 mm	0.20	65 Km/h
1 mm	0.42	96 Km/h
0.5 mm	0.10	65 Km/h
0.5 mm	0.21	96 Km/h

4.10 0 Data transfer system in ASFT

All friction data is created in the Measure Computer (1) sent via Bluetooth to the Presentation/Controller Computer (2) where it is permanently stored and relayed (3) to the Measure Database (4) where every single measure point is backed up and forwarded to any client that is authorized to view it in real time or later (5,6,7) as showed in figure 4.7.

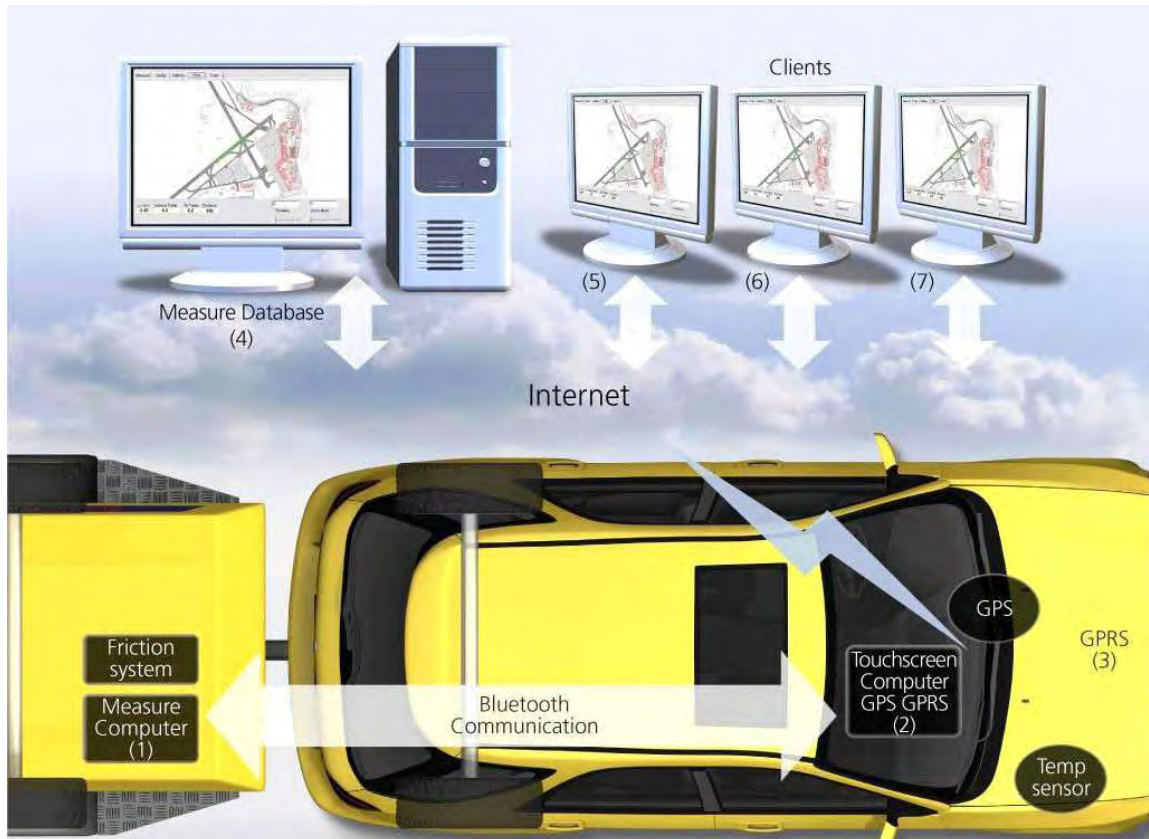


Figure 4.7: Technique of Data Transfer System in ASFT.

CHAPTER 5

RUNWAY SURFACE FRICTION DATA COLLECTION AND ANALYSIS

5.1 Background

Shah Amanat International Airport is the second largest international airport of Bangladesh situated in the southeastern port city of Chittagong. An aerial view of Shah Amanat International Airport has been shown in figure 5.1. The airport is located in the Patenga area of the city, 20 kilometres (13 mile 1 NM) west from the city's main commercial hub, GECC Circle and 18.5 kms south of the city's railway station on the north bank of the Karnaphuli River. It is the second largest airport in Bangladesh operated and maintained by the Civil Aviation Authority of Bangladesh. It is also used by the Bangladesh Air Force.



Figure 5.1: Aerial View of Shah Amanat International Airport, Chittagong.

It was formerly known as MA Hannan International Airport, named after Awami League politician M.A. Hannan, but was renamed on 2 April 2005 by the government of Bangladesh, after an Islamic saint Hazrat Shah Amanat.

5.2 Brief History of Shah Amanat International Airport

The airfield was built in the early 1940s during the British reign. Known as Chittagong Airfield during World War II, the airport was used as a combat airfield, as well as a supply point and photographic reconnaissance base by the United States Army Air Forces Tenth Air Force during the Burma Campaign 1944-

1945. It officially became a Bangladeshi airport in 1972 after Bangladesh's liberation war. At first, it was mainly used for connecting Dhaka and Chittagong, but in the mid-

1990s Biman started international flights to Dubai and a few Saudi Arabian cities and the airport officially became an international airport. In March 1998, a major renovation and expansion began at the airport, which ended in December 2000. CAAB received financial assistance from the Japan International Cooperation Agency for the US\$51.57 million upgrade. The project was carried out by Japanese firms Shimizu & Marubeni. The upgradation project modernized the terminal with new and better seats, more check-

in counters, better aviation security equipment and other facilities. The Air Traffic Control tower also received new hi-

tech equipments such as 3D RADARs. The runway, taxiways and the apron were expanded and improved. After the upgrade, aircrafts such as the Boeing 747-

400 or the Airbus A340 can land easily at the airport. In 2010s Emirates Sky Cargo launched cargo services in 2013, making it the first scheduled cargo airline in the airport. Besides these several airlines are highly interested for operating Cargo flight from this airport.

5.3 General Description of Shah Amanat International Airport

Shah Amanat International Airport, is an international airport serving Bangladesh's southern port city of Chittagong as shown in figure 5.2. Operated and maintained by the Civil Aviation Authority of Bangladesh, it is the second largest airport in

Bangladesh. Shah Amanat International Airport, located in Chittagong has a single runway, which is 9646 feet (2940 metres) long. The geographic coordinates of this airport are 22 degrees, 14 minutes, 59 seconds north (22.249611) and 91 degrees, 48 minutes, 48 seconds east (91.813286). Shah Amanat International Airport is 12 feet (4 m) above sea level. The airport is capable of annually handling 1.5 million passengers and 6,000 tons of cargo. The airport is referred to by the International Air Transport Association (IATA) using the airport code CGP. The International Civil Aviation Organization (ICAO) uses VGGG when referring to Shah Amanat International Airport.

(a) Terminal

The airport's sole 220,000 square feet (20,000 m²) Passenger terminal is divided into two parts: International and Domestic with a boarding bridge in each. The International part of the terminal is larger than the Domestic one due to a higher number of passengers. The building is also divided into two floors: The lower floor is used for checking in, boarding or getting off small planes and receiving luggage while the upper floor is used for boarding or getting off large planes only. The airport also has only 29,063 square feet (2,700 m²) cargo terminal without Cargo Aircraft Parking Apron.

(b) Control Tower

The airport's air traffic control tower is 50 meters west of the airport terminal. It has a clear view of the apron and taxiways but is far from the runway. Heavy rain or fog can make it difficult for controllers to observe planes taking off or landing.

(c) Runway

The airport has a single runway (05/23), which is 2,940 m × 45 m (9,646 ft × 148 ft) as shown in figure 5.2. The largest aircraft that can land in the airport is a Boeing 777-300ER.

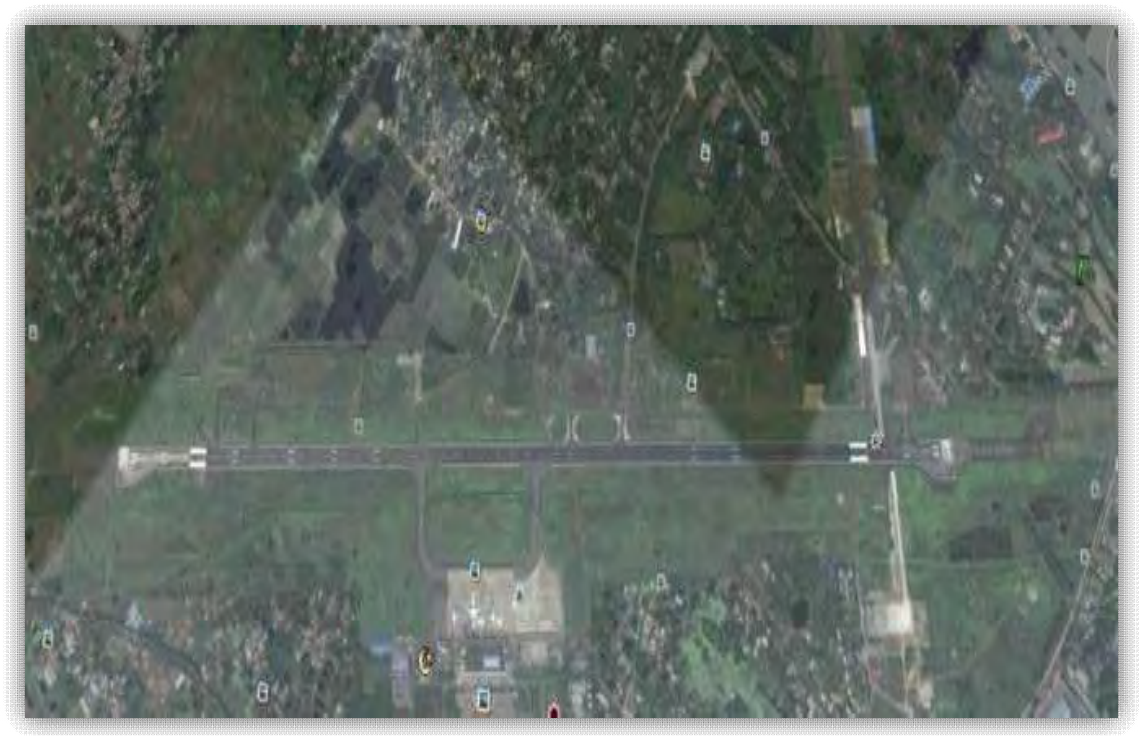


Figure 5.2: Runway of Shah Amanat International Airport, Chittagong.

(d) Taxiways and Apron

The airport has two taxiways, Alpha and Bravo, that directly lead to the apron or aircraft parking zone, from the runway. The apron can accommodate a maximum of four aircraft; two wide-body Boeing 747-400s, a wide-body McDonnell Douglas DC-10 and a narrow-body Airbus A320 can be parked there at once. The airport has two boarding bridges and two passenger steps. The parking points are usually empty as most of the planes that arrive there take off soon after and the planes of local airlines are generally parked at Shahjalal International Airport overnight. A small civil plane hangar belonging to Bangladesh Biman is available but is rarely used. The Bangladesh Military has a parking zone and two plane hangars east of the runway. The Bangladesh Air Force stores a few planes here which have direct access to the runway.



Figure 5.3: Airside View of Shah Amanat International Airport (SAIA).

(e) Aerodromes

The airport has two longitudinal canals parallel to the runway on either side. The side strip at the bottom side of the runway is 150m x 2940m as per ICAO standard for an international airport. The strips are properly graded and maintained as per an aerodrome maintenance manual. The airport has the provision of parallel taxiways to enhance the capacity in the future.

5.4 Data Collection

Shah Amanat International Airport, Chittagong has a single runway which is made of asphalt concrete. The location of the airport has been shown in figure 5.4. The northern end of this runway is designated as RWY 23 and the southern end is designated as RWY

05. Surface friction of the runway has been measured by ASFT car following the rules and regulations ratified by International Civil Aviation

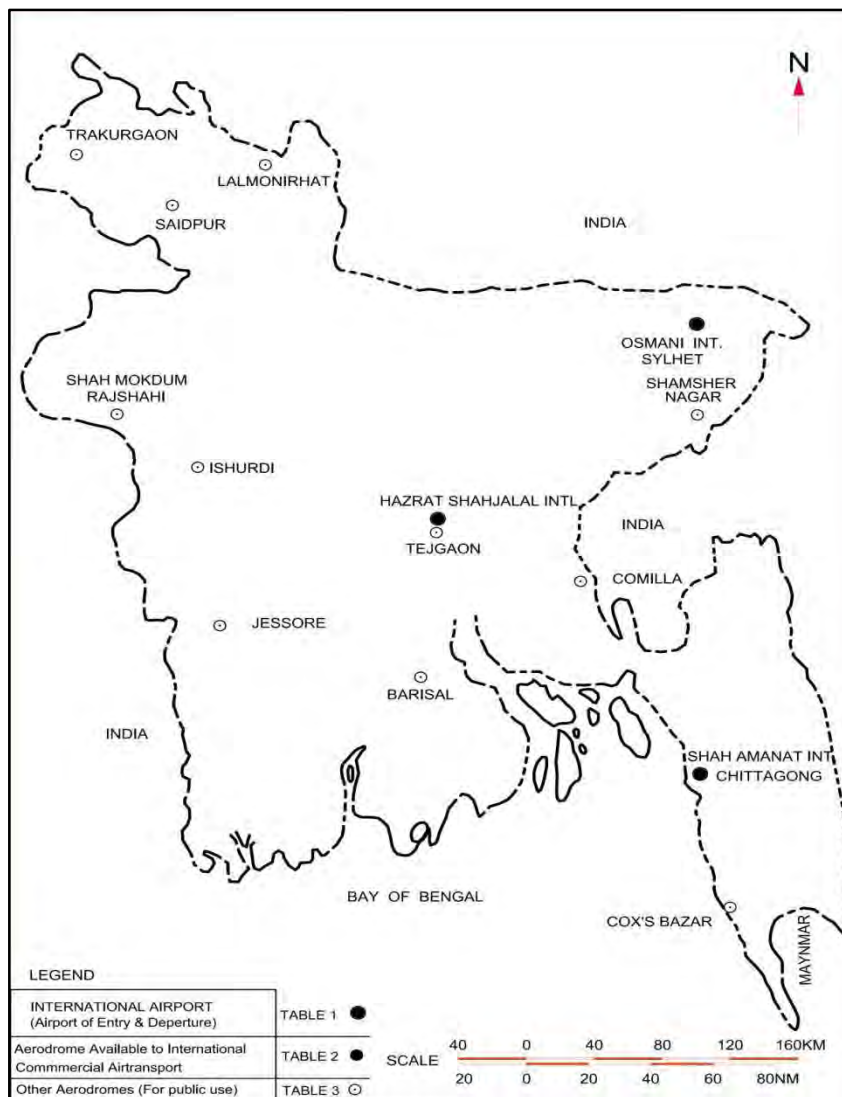


Figure 5.4: Location Map of Shah Amanat International Airport (SAIA).

Organization. The tire type was selected T520 which is made of natural rubber and has a high air pressure (7 bar or 100 psi) as shown in figure 5.5. It's a ribbed tire with three grooves and similar to aircraft tire. The system should be calibrated before it is used. It is of big importance that this is done with accuracy, an incorrectly calibrated system will never give any trustworthy results. Calibration of friction, load and distance were

accomplished according to ASFT manuals. The water pressure was also calibrated because it is significant for self-watering friction measuring device.



Figure 5.5: Tire type T520.

Friction coefficient value was taken from a single run of the car from one end to the other end of the runway maintaining the procedures of ASFT company which is also approved by ICAO by CAAB team as shown in figure 5.6.



Figure 5.6: Runway Surface Friction Measurement Team of CAAB

The speed of the friction measuring car was specified by the ASFT and ICAO. The measured data have been tabulated in the following tables and ppc generated graphs have been illustrated in the following figures.

Table 5.1: FrictionMeasure Report 1.

Configuration	RWY05	TireType	T520
Dateand Time	15-03-14 12:21:44	WaterFilm	ON
Type	ICAO	Average Speed	100km/hr
Equipment	SFT0746		
Pilot	Parimal		
IceLevel	0.25		
RunwayLength	2500		
Location	CHITTAGONG		

Table 5.2: ResultSummary offirst measurement.

Runway	Fric. A	Fric.B	Fric. C	Fric.Max	Fric.Min	Fric avg	Ice
RWY05	0.52 μ	0.65 μ	0.68 μ	0.76 μ	0.29 μ	0.62 μ	0.00%
RWY23	0.54 μ	0.68 μ	0.69 μ	0.80 μ	0.30 μ	0.64 μ	0.00%

Table 5.3: RunwayFrictionCoefficientfromfirstmeasurement.

RW	Fric. A	Fric.B	Fric. C	Fric.Max	Fric.Min	Fric avg	Ice
All	0.53	0.67	0.69	0.80	0.29	0.63	0.00%

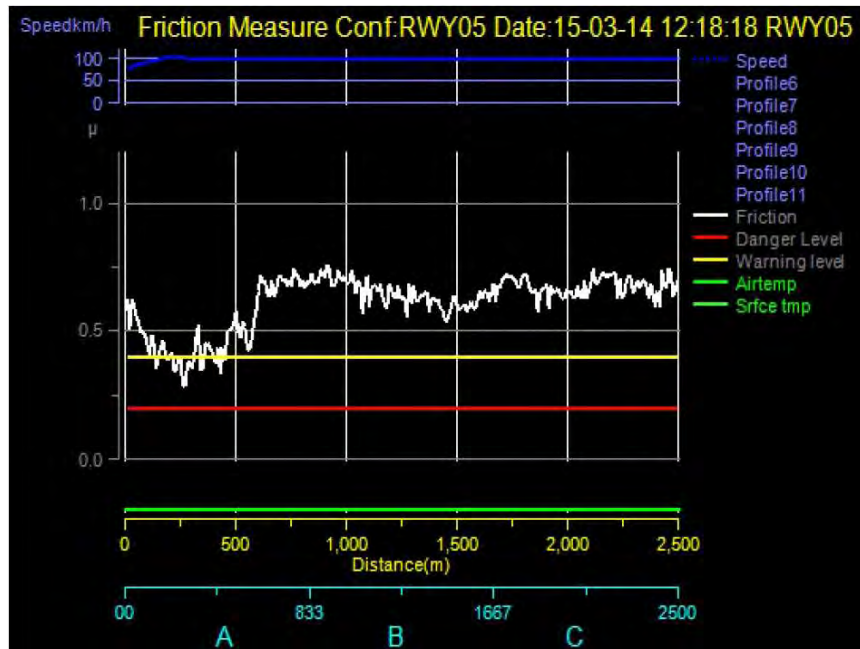


Figure 5.7:FrictionMeasurementGraph from RWY05to RWY23.

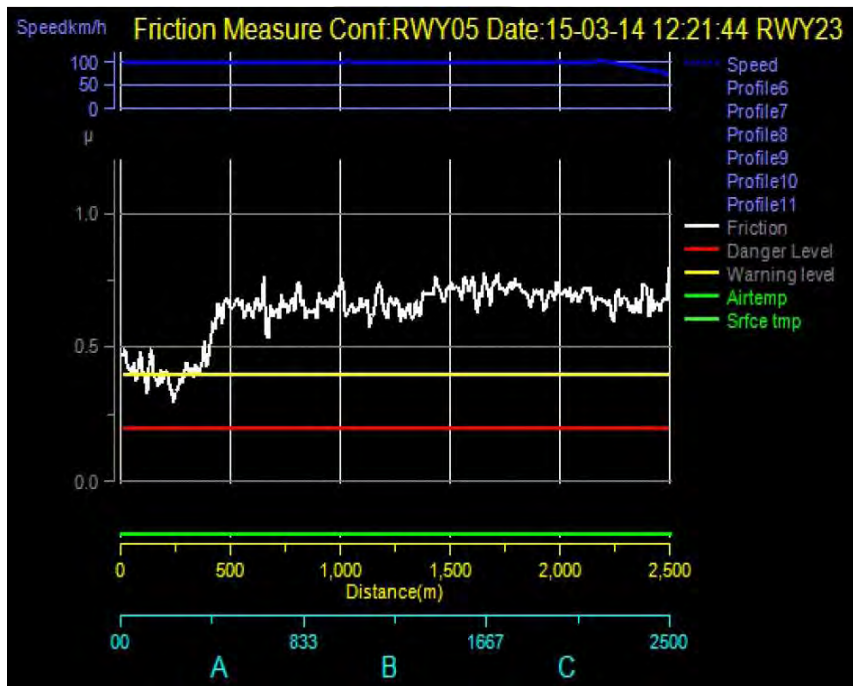


Figure 5.8:FrictionMeasurementGraph from RWY23 toRWY05.

Table 5.4: FrictionMeasure Report 2.

Configuration	RWY05	TireType	T520
Dateand Time	15-03-14 12:41:58	WaterFilm	ON
Type	ICAO	Average Speed	98km/hr
Equipment	SFT0746		
Pilot	Parimal		
IceLevel	0.25		
RunwayLength	2500		
Location	CHITTAGONG		

Table 5.5: ResultSummary ofsecondmeasurement.

Runway	Fric. A	Fric.B	Fric. C	Fric. Max	Fric.Min	Fricavg	Ice
RWY05	0.51 μ	0.63 μ	0.68 μ	0.76 μ	0.28 μ	0.61 μ	0.00%
RWY23	0.51 μ	0.67 μ	0.69 μ	0.85 μ	0.27 μ	0.62 μ	0.00%

Table 5.6: RunwayFrictionCoefficientfromsecond measurement.

RW	Fric. A	Fric.B	Fric. C	Fric.Max	Fric.Min	Fric avg	Ice
All	0.51	0.65	0.69	0.85	0.27	0.61	0.00%

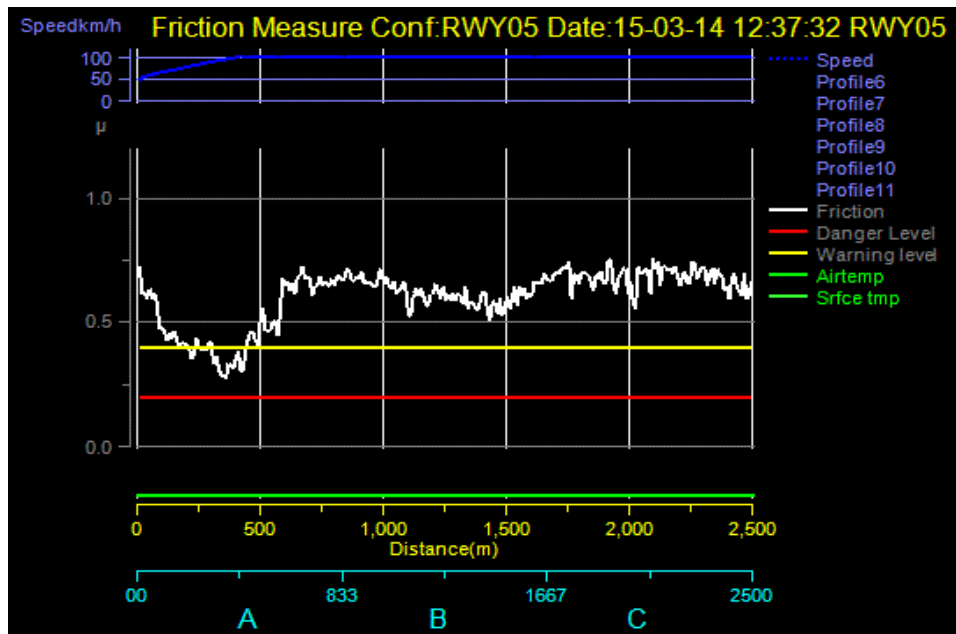


Figure 5.9:FrictionMeasurementGraph from RWY05 toRWY23.

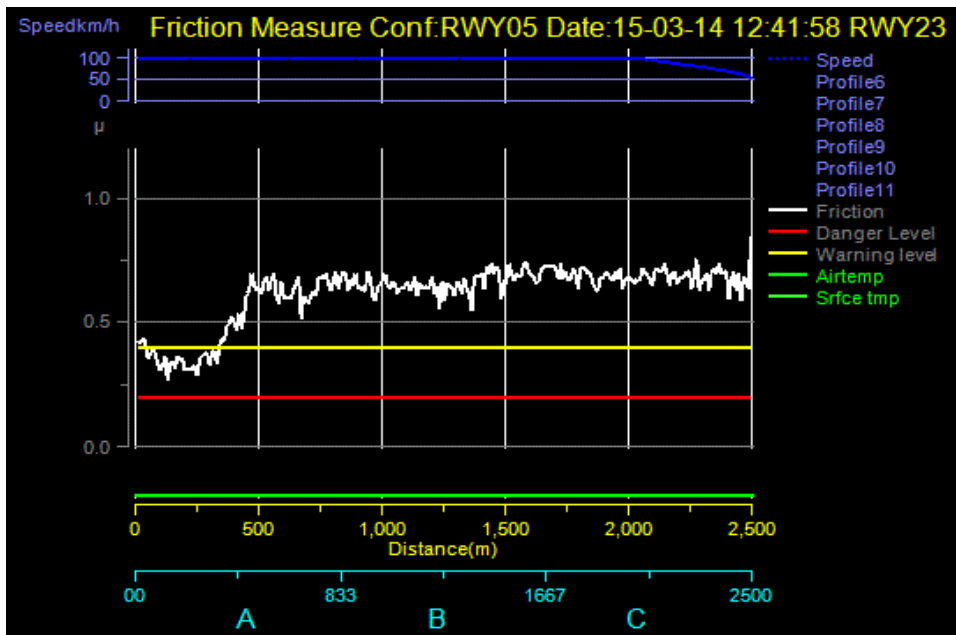


Figure 5.10:FrictionMeasurementGraph from RWY05 toRWY23.

5.5 Reporting of aircraft movement surface condition

An aircraft movement surface condition report can be presented by a friction coefficient number along with a surface description and other relevant information. Typically during preflight planning, a NOTAM is available. Once airborne, the crew gets information through the ATIS (Automatic Terminal Information Service) and with rapidly changing conditions, verbal updates are usually available through the tower.

5.6 Development of Runway Surface Friction Coefficient Predicting Model

Runway surface friction coefficient predicting model will be developed applying Multiple Linear Regression Analysis Technique. Multiple regression is a logical extension of the principles of simple linear regression to situations in which there are several predictor variables. For instance if we have two predictor variables, X_1 and X_2 , then the form of the model is given by:

$$Y = b_0 + b_1X_1 + b_2X_2 + \dots + \epsilon \dots \dots \dots (5.1)$$

which comprises a deterministic component involving the three regression coefficients (b_0, b_1 and b_2) and a random component involving the residual (error) term. The error term is unknown because the true model is unknown. Once the model has been estimated, the regression residuals are defined as the difference between the observed and predicted values. The residuals measure the closeness of the predicted values. The algorithm for estimating the regression equation (solution of the normal equations) guarantees that the residuals have a mean of zero. The variance of the residuals measures

the “size” of the error, and is small if the model fits the data well. In this respect, the Multiple Linear Regression equation will be

$$\text{Friction coefficient, } \mu = b_0 + b_1 \times \text{Distance} + b_2 \times \text{Speed} + b_3 \times \text{air temperature} + b_4 \times \text{ground temperature} + b_5 \times \text{tire pressure} \dots \dots \dots (5.2)$$

Analyzing dataset from first measurement (Appendix A) applying regression analysis by MaxStat software, we obtain the numerical values of the regression coefficients b_0, b_1, b_2, b_3, b_4 and b_5 tabulated in table 5.10.

Table 5.10: Statistical Analysis of Regression coefficients.

Regression coefficients	Magnitude	CI Lower 95%	CI Upper 95%
Intercept, b_0	0.6623		
Distance passed, b_1	-0.0001167	-0.000166	-0.000067
Speed, b_2	+0.01405	0.0038	0.0242
Air temperature, b_3	+0.0044	-0.051	0.0602
Ground temperature, b_4	-0.00933	-0.038	0.0387
Tire pressure, b_5	-0.16	-0.5477	0.2284

Friction coefficient prediction model equation = 0.6623 -0.0001167 Distance +0.01405 Speed + 0.0044 Air temperature -0.00933 Ground temperature -0.16 Tire pressure.

Goodness of fit of this model can be assessed by R^2 and adjusted R^2 . The explanatory power of the regression is summarized by its R^2 value, also called the coefficient of determination, is often described as the proportion of variance explained by regression. We need to keep in mind that a high R^2 does not imply causation. The R^2 value for a regression can be made arbitrarily high simply by including more and more predictors in the model. The adjusted R^2 is one of several statistics that attempt to compensate for this artificial increase in accuracy. The value of R^2 and adjusted R^2 are respectively 0.80 and 0.749. The numerical values of estimates, t value of t test of the parameters of the regression model and p value have been tabulated in table 5.11.

Table 5.11 : Statistics of Surface Friction Prediction Model.

Parameter	Estimate	t _{critical}	t	Pvalue
Intercept	0.6623			< 0.0001
Distance	-0.0001167	2.064	8.829	< 0.0001
Speed	+0.01405	2.064	65.060	< 0.0001
Air temperature	+0.0044	2.069	31278.7	< 0.0001
Ground Temperature	-0.00933	2.069	125.292	< 0.0001
Tire pressure	-0.16	2.069	273.372	< 0.0001

Linear regression of μ predicted (y) vs μ observed(x)

Graphical relationship between predicted μ and observed μ has been shown in figure 5.13.

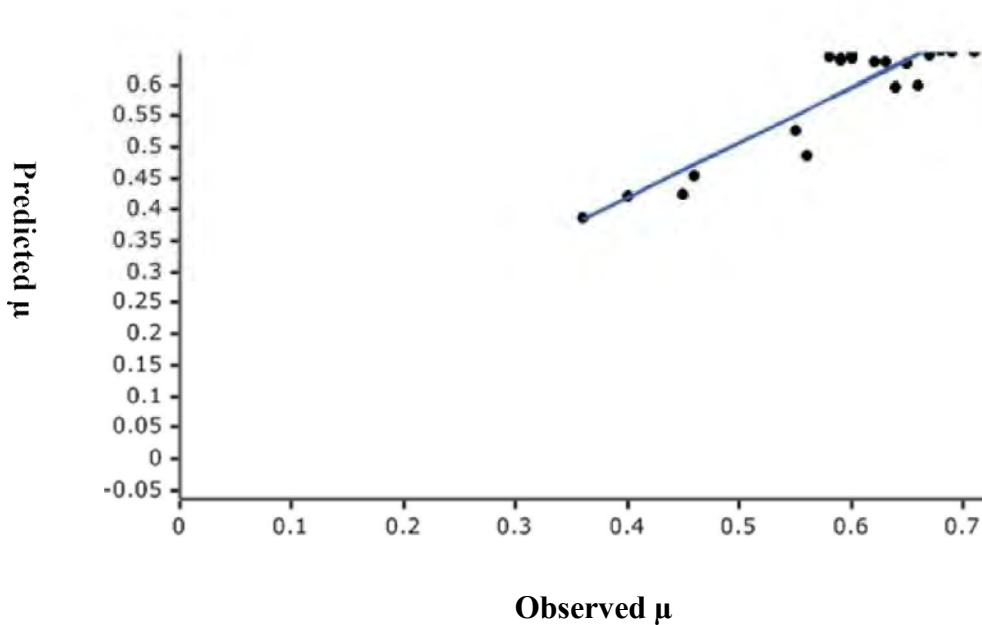


Figure 5.13: Correlation between Predicted μ and Observed μ .

Applying regression analysis by MaxStat software, we obtain the numerical values of the regression coefficients b_0 , b_1 .

$b_0 = 0.11$ (C.I. (95%) of $b_0 \pm 0.099$)

$b_1 = 0.801$ (C.I. (95%) of $b_1 \pm 0.172$)

The numerical values of regression coefficient R^2 and adjusted R^2 are 0.80 and 0.79 respectively.

5.7 Correlation of runway surface friction coefficient with parameters

The Pearson correlation coefficient, r , can take a range of values from +1 to -1. A value of 0 indicates that there is no association between the two variables. A value greater than 0 indicates a positive association, that is, as the value of one variable increases so does the value of the other variable. A value less than 0 indicates a negative association, that is, as the value of one variable increases the value of the other variable decreases. Values for r between +1 and -1 (for example, $r = 0.8$ or -0.4) indicate that there is variation around the line of best fit. The closer the value of r to 0 the greater the variation around the line of best fit. The above statistical analysis stipulates that there is a positive association between runway surface friction coefficient and speed. Pearson Correlation Coefficient has been determined from Runway surface friction coefficient vs Speed graph. The value of Pearson Correlation Coefficient is 0.76. Confidence interval (95%) of r is lower CI is 0.51 and upper CI is 0.89. With the confidence interval we can express the precision of the mean with a defined probability (usually 95%). The confidence interval depends on the sample size and the variability (standard deviation). The 95% confidence interval lies between 0.51 and 0.89, i.e. the real mean lies in this range with a probability of 95%. This Pearson Correlation Coefficient value of air temperature, ground temperature, distance and tire pressure from table 5.1 with surface friction coefficient has also been determined and corresponding lower CI, upper CI and

p value have been tabulated in the following tables and graphical relationships have been illustrated in the following figures.

Table 5.12 : Correlation Statistics of Friction Coefficient and Speed.

Correlation coefficient	Lower CI 95%	Upper CI 95%	P value
0.76	0.51	0.89	0.1639

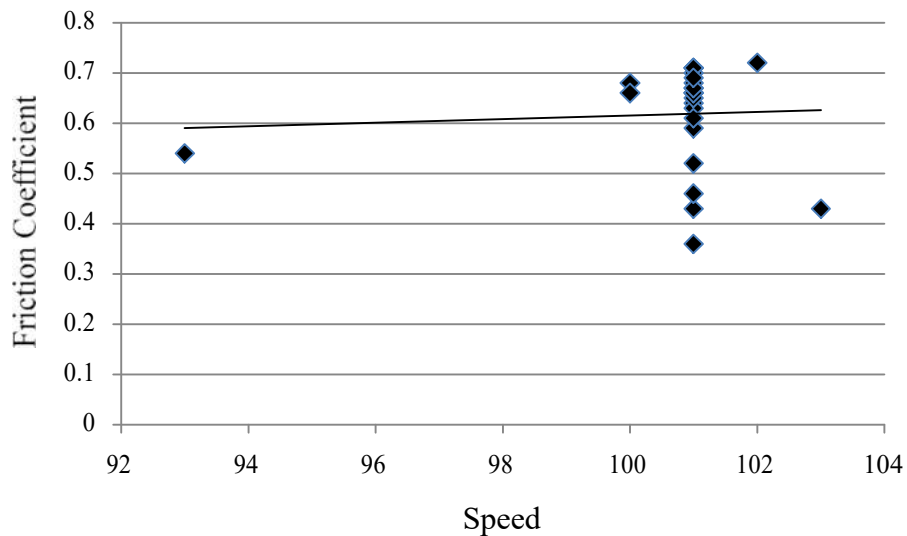


Figure 5.14: Correlation between Friction Coefficient and Speed.

Correlation of runway surface friction coefficient with air temperature

Table 5.13 : Correlation Statistics of Friction Coefficient and Air Temperature.

Correlation coefficient	Lower CI 95%	Upper CI 95%	P value
0.337	-0.008	0.610	0.1074

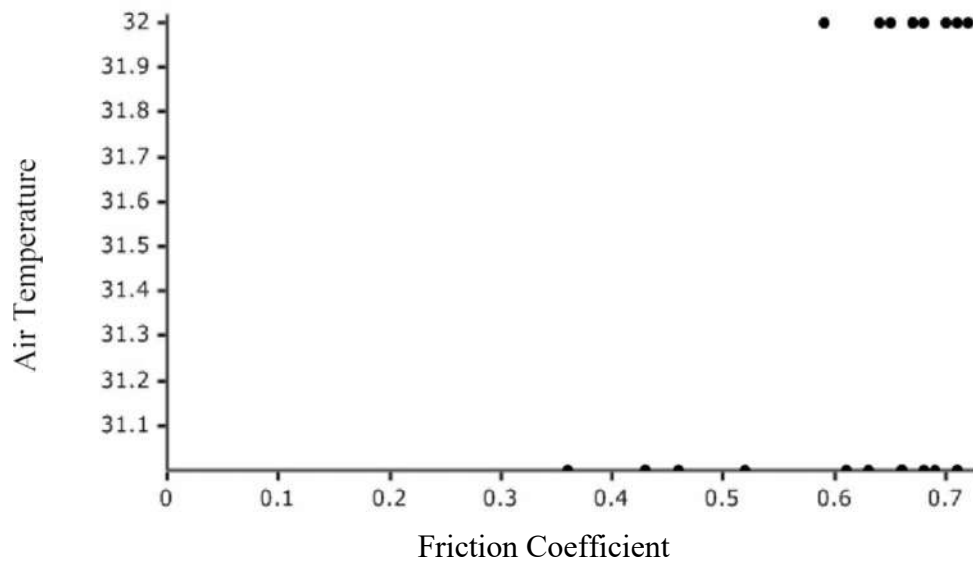


Figure5.15: Correlation betweenFriction Coefficientand Air Temperature.

Correlationof runway surfacefriction coefficient withdistance

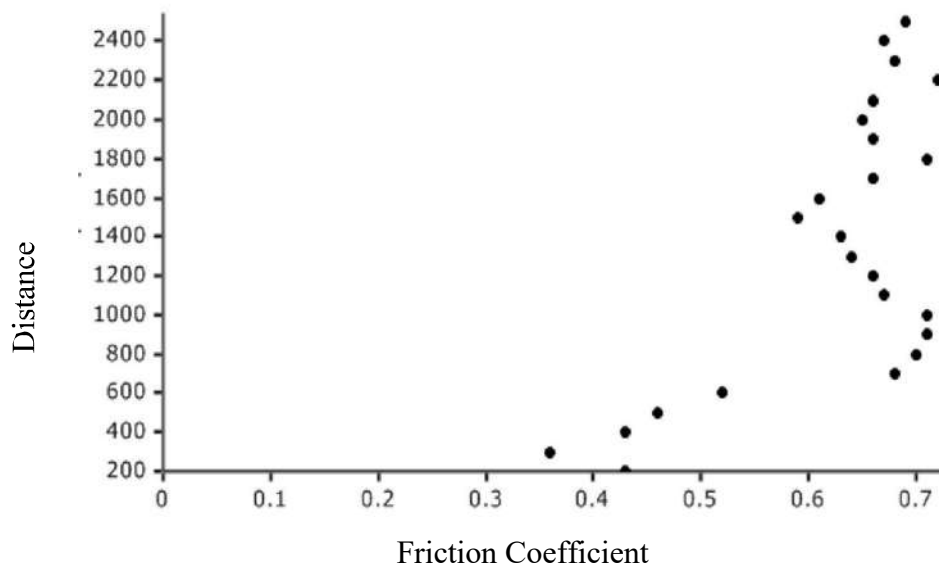


Figure5.16: Correlation betweenFriction CoefficientandDistance.

Table 5.14 : Correlation Statistics of Friction Coefficient and Distance.

Correlation coefficient	Lower CI 95%	Upper CI 95%	P value
0.661	0.352	0.840	0.0004

Correlation of runway surface friction coefficient with tire pressure

Table 5.15 : Correlation Statistics of Friction Coefficient and Tire Pressure.

Correlation coefficient	Lower CI 95%	Upper CI 95%	P value
-0.228	-0.578	0.193	0.2834

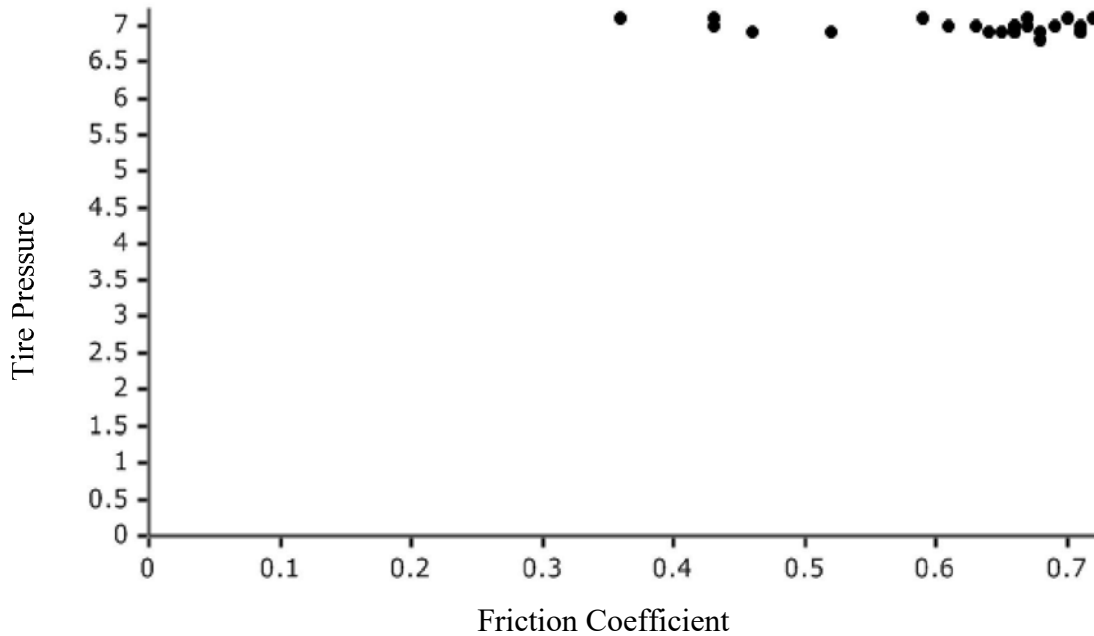


Figure 5.17: Correlation between Friction Coefficient and Tire Pressure.

Correlation of runway surface friction coefficient with ground temperature

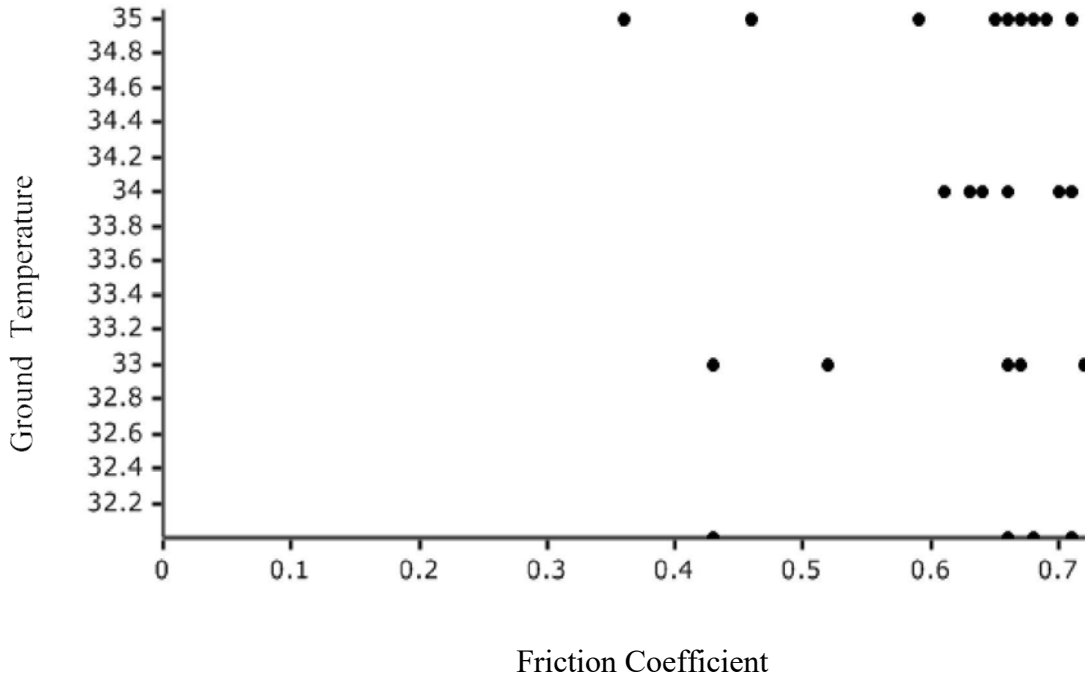


Figure 5.18: Correlation between Friction Coefficient and Ground

Temperature. Table 5.16 : Correlation Statistics of Friction Coefficient and Ground Temperature.

Correlation coefficient	Lower CI 95%	Upper CI 95%	P value
-0.003	-0.405	0.401	0.9907

5.8 Sensitivity Analysis of different parameters of surface friction coefficient

The most significant parameter influencing surface friction coefficient has been identified by increasing numerical value of every single parameter by 20% and considering remaining parameter constant. This analysis has been done for each parameter. The numerical value of each parameter for this analysis has been taken from appendix B. This sensitivity analysis has been tabulated

in the table 5.17.

Table 5.17 : Sensitivity Analysis of surface friction coefficient prediction model.

Parameter	μ value from model equation	μ value after 20% increment of parameter	Remarks
Distance	0.68	0.66	2.9% decrement
Speed	0.68	0.96	42% increment
Ground Temperature	0.68	0.62	8.8% decrement
Air temperature	0.68	0.71	4.4% increment
Tire pressure	0.68	0.46	32.6% decrement

5.9 Terminology

R^2

The R^2 measures the variation the “dependent” variable that can be accounted for or “explained” by independent variables.

Adjusted R^2

If the number of variables is a significant fraction of the number of observations, then the problem of over-fitting occurs. The adjusted R^2 is a measure of the noise around the regression line, correcting for this over-fitting problem.

t test

A paired t test is used to compare means of two data groups and where observations in one group can be paired with observations in the other group.

Pearson Correlation Coefficient

Similar to linear regression, correlation determines the linear relationship between two variables, but neither is assumed to be functionally dependent on each other. It is most important that correlation does not mean causation.

p value

The p value can be interpreted in terms of a hypothetical repetition of the study. Assuming the null hypothesis true and a new dataset is obtained independently of the first dataset but using the same sampling, then the probability of the new value confirming the original value is p value.

Chapter 6

CONCLUSIONS AND RECOMMENDATIONS

6.1 Conclusion

In this paper, wet and contaminated condition of runway surface has been taken into consideration in developing runway surface friction coefficient prediction model. The key findings of this paper are as follows:

- Available braking friction coefficient is ground speed dependent. With the speed increase the available braking friction coefficient decreases.

In order to be able to explain the phenomenon of frictional variation due to temperature theoretically, one needs to understand the properties of the two types of materials that interact in producing pavement friction. The stiffness of both materials in contact, rubber and asphalt, are expected to decrease with increasing temperature.

The decrease of coefficient of friction with the increase in water temperature could be attributed to the sensitivity of the hydrodynamic properties of water to temperature. The viscosity of water decreases as temperature increases thereby decreasing the boundary layer shear stress. This is because the shear stress in a Newtonian fluid is equal to the product of the viscosity and the time rate of strain.

- The presence of contaminants on the runway reduces the friction between the tire and runway surface. The reduction is a function of several factors including the tire-pavement interaction, the anti-locking system performance, type of runway pavement, and the type of contaminants also.

- The centred best fit straight line in friction coefficient vs speed graph stipulates that wet runway remains a good runway friction condition which is similar to the dry runway pavement findings.
- The results indicate that when the speed is low, the wet runway has a maximum available braking friction that is nearly the same as the dry runway. With the speed increases, the maximum available braking friction increases for both wet runways and dry runways. However, a bigger drop in maximum available braking friction occurs when the runway is wet. An increment of speed value by 20% deduces 42% increment of coefficient of surface friction value of runway surface.

The cruise speed of the ASFT car was not in wider range. Therefore, the graph has not demonstrated a distinct relationship between speed and friction coefficient. Pearson correlation coefficient found from statistical analysis is 0.337. From the regression equation, we can conclude that with the increase of speed, friction coefficient value will increase and vice versa.

The results reveal that distance parameter shows a negative association with friction coefficient of runway surface. An increment of distance value by 20% simultaneously decreases numerical value of friction coefficient by 2.9%.

The regression equation stipulates that when the ground temperature is high, the friction coefficient will decrease consequently. Friction measurement was accomplished in a day. Therefore, there was no variation in ground temperature data points and Pearson correlation coefficient was found very insignificant. It is found that an increment of ground temperature value of 20% is associated with a 8.8% decrement of friction coefficient value of runway surface.

The results reveal that air temperature parameters show a positive association with the friction coefficient of runway surface. An increment of air temperature value by 20% increases friction coefficient value of runway surface by 4.4%.

The results reveal that when tire pressure increases, friction coefficient of runway surface decreases. During the data collection period, the tire pressure was already same. Consequently, variation of friction coefficient with tire pressure did not show a significant relationship. Pearson correlation coefficient has showed a negative association between tire pressure and friction coefficient. It is found that an increment of tire pressure value by 20% decreases friction coefficient value of runway surface by 32.6%.

6.2 Future Scope of the Study

In this research, runway roughness is not taken into consideration. However, its influence on runway surface friction coefficients should be conducted in the future study.

Full braking or maximum braking landing test of commercial aircraft are also recommended to validate the surface friction coefficient prediction model.

Shah Amanat International Airport maintains its runway rapidly in a good condition, hydroplaning and insufficient friction braking due to contaminants did not occur in the collected data. Future aircraft landing test on runways with severe wet and contaminated conditions are recommended to validate the model.

This research can be used as a significant tool for developing a model for predicting aircraft landing distance irrespective of aircraft type as a function of runway surface friction coefficient.

This research will help to identify possibilities of harmonizing runway friction characteristic measurement technologies and provide a basis for improving and harmonizing the implementation of current ICAO Standards and Recommended Practices (SARPS). This could provide the opportunity for a global standardized application, and contribute to the progress of the ICAO action plan.

The results retrieved from this research are ready for discussion with ICAO working groups, experts and the stakeholder communities but may also be reviewed in the light of the work carried out by the FAA Takeoff and Landing Performance Assessment - Aviation Rulemaking Committee (TALPA/ARC).

By adopting a systematic approach to the measurement of runway surface friction characteristics, the degradation of runway surface friction can be determined by the comparison and assessment of data over time. By utilising this data, aerodrome operators should be in a position to target maintenance as required in order to help ensure aircraft braking performance does not fall below internationally accepted levels.

6.3 Recommendations

The friction characteristics of a runway will vary over time as the runway is subject to wear and tear (polishing), accumulation of rubber deposits and to the effects of weather and other environmental conditions. Aerodrome operators should monitor the results of assessments and should alter the interval between assessments depending on the results. If historical data indicate that the surface is deteriorating relatively quickly, more frequent monitoring may be required in order to ensure that maintenance is arranged before the friction characteristics deteriorate to MFL. The aerodrome operator should record the justification for any variation from the recommended periodicity for assessments.

When there are indications that the friction characteristics of a runway may be reduced because of poor drainage, an additional assessment should be conducted, but this time under natural conditions representative of local rain. This assessment differs in that water depths in the poorly drained areas are normally greater in local rain conditions. The results are thus more appropriate to identify problem areas having low friction values that could induce hydroplaning than the standard assessment method. If circumstances do not permit assessments to be conducted during natural conditions representative of rain, then doubling the runway surface with water may simulate this condition.

When conducting assessments on wet runways, it is important to note that, unlike compacted snow and ice conditions, in which there is very limited variation of the friction reading with speed, a wet runway produces a drop in friction with an increase in speed. However, as the speed increases, the rate at which the friction is reduced becomes less. Among the factors affecting friction between the tire and the runway surface, texture is particularly important. If the runway has a good macro-texture (roughness) allowing the water to escape beneath the tire, then the friction value will be less affected by speed. Conversely, a low macro-texture (smooth) surface will produce a larger drop in friction as speed increases.

CFME manufacturers should be consulted concerning any special operating procedures involving testing at high speeds. Operational safety assessments relating to specific aerodrome procedures may need to be reviewed to take into account testing at high speeds.

The CFME operators should ensure that the equipment is in full working order and calibrated in accordance with the manufacturers' operating instructions. Those with responsibility for the provision of CFME should ensure that the equipment is serviced regularly and that the measuring tire is of the correct

specification and remains within manufacturers' tolerance. General guidance on test speeds, nominal test water film thickness, test tire type, test tire pressure and test tire condition should be sought from the CFME manufacturer.

The success of friction measurement in delivering reliable friction data depends greatly on the personnel who are responsible for operating the CFME. All operators' competent equipment's operation and should be trained and in the maintenance and be aware of the critical factors affecting the accuracy of friction measurements. Training may be conducted during normal assessment runs provided that suitable measures are in place to ensure that the results of the runs are valid. If additional runs are conducted for the purpose of training or maintenance of competence, the results may be included in the assessment system if they are known to be valid.

Aerodrome operators should make effective use of the assessment data produced by CFME. Regular reviews coupled with planned maintenance activities driven by trend analysis will ensure that surface friction characteristics are consistently acceptable. Aerodrome License Holders are recommended to use either CFME manufacturers' software based reporting or to export raw data into an appropriate spreadsheet format. However, detailed examination of the data for each 10m readings should be carried out after each assessment to identify areas of the runway, which may require maintenance or closer monitoring.

On heavily trafficked runways with a prevailing direction of use, CFME operators may detect a difference in results when collecting data on reciprocal runs. Should this be the case the aerodrome operator may wish to seek expert opinion on the implications of any differences recorded.

Dampness, fog and mist conditions might also affect the outcome of the assessment and aerodrome operators should be aware that crosswinds might affect assessments utilizing self-wetting. Aerodrome operators should seek advice on these issues from the CFME manufacturer.

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APPENDIX A

LENGTHWISE FRICTION DATA FROM RWY 05(LAP1).

Distance(m)	Friction	Speed(km/h)	T _{air} (°C)	T _{gnd} (°C)	Tire pressure(bar)
100	0.54	93	32	34	7
200	0.43	103	31	33	7.1
300	0.36	101	31	35	7.1
400	0.43	101	31	32	7
500	0.46	101	31	35	6.9
600	0.52	101	31	33	6.9
700	0.68	100	32	32	6.9
800	0.70	101	32	34	7.1
900	0.71	101	32	35	6.9
1000	0.71	101	31	32	7
1100	0.67	101	32	33	7
1200	0.66	101	31	34	7
1300	0.64	101	32	34	6.9
1400	0.63	101	31	34	7
1500	0.59	101	32	35	7.1
1600	0.61	101	31	34	7
1700	0.66	100	31	33	6.9
1800	0.71	101	31	34	6.9
1900	0.66	101	31	35	7
2000	0.65	101	32	35	6.9
2100	0.66	101	31	32	6.9
2200	0.72	102	32	33	7.1
2300	0.68	101	31	35	6.8
2400	0.67	101	32	35	7.1
2500	0.69	101	31	35	7

APPENDIXB
LENGTHWISE FRICTION DATA FROM RWY23(LAP 2).

Distance(m)	Friction	Speed(km/h)	T _{air} (°C)	T _{gnd} (°C)	Tire pressure(bar)
100	0.54	85	32	34	7
200	0.68	94	31	33	7.1
300	0.67	102	31	35	7.1
400	0.69	101	31	32	7
500	0.70	101	31	35	6.9
600	0.70	101	31	33	6.9
700	0.71	101	32	32	6.9
800	0.73	101	32	34	7.1
900	0.71	101	32	35	6.9
1000	0.73	101	31	32	7
1100	0.66	101	32	33	7
1200	0.65	101	31	34	7
1300	0.67	101	32	34	6.9
1400	0.67	101	31	34	7
1500	0.67	101	32	35	7
1600	0.68	101	31	34	7
1700	0.66	101	31	33	6.9
1800	0.65	101	31	34	6.9
1900	0.64	101	31	35	7
2000	0.66	101	32	35	6.9
2100	0.61	101	31	32	6.9
2200	0.43	101	32	33	7.1
2300	0.37	101	31	35	6.8
2400	0.40	101	32	35	7.1
2500	0.45	101	31	35	7

APPENDIXC
LENGTHWISE FRICTION DATA FROM RWY05 (LAP 3).

Distance(m)	Friction	Speed(km/h)	T _{air} (°C)	T _{gnd} (°C)	Tire pressure(bar)
100	0.60	69	32	34	7
200	0.44	80	31	33	7.1
300	0.40	92	31	35	7.1
400	0.32	102	31	32	7
500	0.41	102	31	35	6.9
600	0.53	103	31	33	7
700	0.67	102	32	32	6.89
800	0.66	103	32	34	7.1
900	0.69	103	32	35	6.9
1000	0.67	103	31	32	7
1100	0.64	102	32	33	7
1200	0.61	102	31	34	7
1300	0.62	103	32	34	6.9
1400	0.59	103	31	34	7
1500	0.57	103	32	35	7
1600	0.61	102	31	34	7
1700	0.68	103	31	33	6.9
1800	0.70	103	31	34	6.9
1900	0.68	102	31	35	7
2000	0.69	102	32	35	6.9
2100	0.68	103	31	32	6.9
2200	0.71	103	32	33	7.1
2300	0.71	102	31	35	6.8
2400	0.67	103	32	35	7.1
2500	0.65	103	31	35	7

APPENDIXD

LENGTHWISE FRICTION DATA FROM RWY23(Lap4).

Distance(m)	Friction	Speed(km/h)	T _{air} (°C)	T _{gnd} (°C)	Tire pressure(bar)
100	0.68	70	32	34	7
200	0.70	79	31	33	7.1
300	0.69	87	31	35	7.1
400	0.70	95	31	32	7
500	0.68	101	31	35	6.9
600	0.69	100	31	33	7
700	0.67	100	32	32	6.9
800	0.66	100	32	34	7.1
900	0.69	100	32	35	6.9
1000	0.67	100	31	32	7
1100	0.64	100	32	33	7
1200	0.61	100	31	34	7
1300	0.62	100	32	34	6.9
1400	0.59	100	31	34	7
1500	0.57	100	32	35	7
1600	0.61	99	31	34	7
1700	0.68	100	31	33	6.9
1800	0.70	100	31	34	6.89
1900	0.68	100	31	35	6.9
2000	0.69	100	32	35	7
2100	0.68	100	31	32	6.9
2200	0.71	100	32	33	7.1
2300	0.71	99	31	35	6.9
2400	0.67	99	32	35	7.1
2500	0.65	100	31	35	7

APPENDIXE
LENGTHWISE FRICTION DATA FROM RWY23(LAP 5).

Distance(m)	Friction	Speed(km/h)	T _{air} (°C)	T _{gnd} (°C)	Tire pressure(bar)
100	0.56	97	32	34	7
200	0.46	98	31	33	7.1
300	0.40	99	31	35	7.1
400	0.36	100	31	32	7
500	0.45	99	31	35	6.9
600	0.55	96	31	31	6.9
700	0.64	94	32	32	6.9
800	0.66	94	32	34	7.1
900	0.65	93	32	35	6.9
1000	0.63	93	31	32	7
1100	0.62	93	32	33	7
1200	0.59	93	31	34	7
1300	0.59	93	32	32	7
1400	0.60	93	31	34	7
1500	0.58	93	32	35	7
1600	0.60	93	31	34	7
1700	0.67	93	31	33	6.9
1800	0.60	93	31	34	6.9
1900	0.71	93	31	35	6.9
2000	0.69	93	32	35	6.9
2100	0.68	93	31	32	6.9
2200	0.68	93	32	33	7.1
2300	0.68	93	31	35	6.9
2400	0.72	93	32	35	7.1
2500	0.72	93	31	35	7

APPENDIX F
LENGTHWISE FRICTION DATA FROM RWY05(LAP 6).

Distance(m)	Friction	Speed(km/h)	T _{air} (°C)	T _{gnd} (°C)	Tire pressure(bar)
100	0.63	87	32	34	7
200	0.66	87	31	33	7.1
300	0.71	87	31	35	7.1
400	0.72	87	31	32	7
500	0.73	87	31	35	6.9
600	0.70	87	31	31	6.9
700	0.70	87	32	32	6.9
800	0.68	87	32	34	7.1
900	0.71	87	32	35	6.9
1000	0.71	87	31	32	7
1100	0.71	87	32	33	7
1200	0.72	87	31	34	6.9
1300	0.71	87	32	32	7
1400	0.69	87	31	34	6.9
1500	0.66	87	32	35	7
1600	0.63	87	31	34	7
1700	0.67	87	31	33	6.9
1800	0.64	87	31	34	7
1900	0.70	87	31	35	7
2000	0.69	87	32	35	7.1
2100	0.58	87	31	32	6.9
2200	0.49	87	32	33	7.1
2300	0.48	87	31	35	6.9
2400	0.47	87	32	35	7.1
2500	0.72	93	31	35	7

APPENDIX G

HAZARDS RELATED TO FRICTION ISSUES AND PAVEMENT

Hazard	Friction characteristics			Significant change
	Physical	Functional	Operational	
	Microtexture	Slippery	Slippery	Retexture
	Macrottexture	Wet smooth		BC
	Macrottexture	Wet skid resistant		DE
No slope	Standing water	Poor drainage at tire/ground interface	Longer stopping distance	New drainage
		Hydroplaning	Loss of directional	
Natural rounded aggregate	Susceptible for polishing	Slippery	Slippery when wet	Retexture Repave
Rubber deposit on crushed aggregate	Cover texture	Reduced texture	No performance credit on Wet skid resistant pavement	Remove Rubber deposit
		Slippery	Slippery	
Rubber deposit on natural, smooth aggregate	Cover texture	Reduced texture Slippery	Longer topping distance , Slippery	
Grooves	Closing due to deformation	Poor drainage at tire/ground interface	Longer topping distance	Open grooves
			No performance credit on Wet skid resistant pavement	
	Filled with contaminant	Poor drainage at tire/ground interface	Longer topping distance	Remove contaminant

APPENDIXH

HAZARDS RELATED TO FRICTION ISSUES AND AIRCRAFT

Hazard	Friction characteristics			Significant change
	Physical	Functional	Operational	
Tire wear	Tire tread depth	Drainage ability at tire/ground interface.	Basic assumption for wet skid resistance	Basic assumption based on tire tread depth of 2mm.
Change in inflation pressure	Inflation pressure	Drainage capability at tire/ground interface.	Basic assumption for wet skid resistance	Curves (e.g. equations) inharmonized certifications specifications for

APPENDIXI

HAZARDS RELATED TO FRICTION ISSUES AND REPORTING FORMAT

Hazard	Friction characteristics			Significant change
	Physical	Functional	Operational	
Clear and dry	Dry	Certification limited	Clear and dry	Dry
Damp	Wet performance data	Damp	Wet performance data	Damp
Wet smooth	Wet	Reduced braking action	Wet performance data	Less than 3 mm
Wet skid resistant	Wet	Reduced braking action	Wet skid resistant performance data	Less than 3 mm
Standing water	Wet	Hydroplaning susceptible	Above 3 mm	Standing water
Rime or frost	Thin layer			

covered	depth normally less than 1 mm			
Loose snow	20 mm ¹			20 mm ¹
Dry snow				
	Coverage	Reduced braking action.	Longer stopping distance	10, 25, 50, 100 per cent
	Depth	Drag force	Longer takeoff distance	20, 40, 60... mm
Wet snow				
	Coverage	Reduced braking action.	Longer stopping distance	10, 25, 50, 100 per cent
	Depth	Drag force	Longer takeoff distance	10, 20, 30... mm
Slush				
	Coverage	Reduced braking action.	Longer stopping distance	10, 25, 50, 100 per cent
	Depth	Drag force	Longer takeoff distance	3, 6, 9, 12 mm
Wet ice				
Compacted snow or ice				
Ice				
	Coverage	Reduced braking action	Longer stopping distance	10, 25, 50, 100 per cent
Compacted or rolled snow	Coverage	Reduced braking action.	Longer stopping distance	10, 25, 50, 100 per cent
Frozen ruts or ridges	Coverage	Reduced braking action.	Longer stopping distance	10, 25, 50, 100 per cent
Sand	Present	Reduced braking action.	Longer stopping distance	
Mud	Present	Reduced braking action.	Longer stopping distance	
Oil/fuel spillage	Present	Reduced braking action.	Longer stopping distance	

APPENDIX J

HAZARDS RELATED TO FRICTION ISSUES AND ATMOSPHERE

Hazard	Friction characteristics			Significant change
	Physical	Functional	Operational	
Precipitation	Contaminant	Influence antiskid	Reduced braking action	
Wind	Crosswind	Move aircraft	Loss of directional	
Temperature	Freezing precipitation	Influence antiskid	Reduced braking action	
Radiation	Freezing moisture on ground	Influence antiskid system	Reduced braking action	